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Bank: (Aviation Mechanic Airframe) Airman Knowledge Test Question Bank

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1. A02A	AMA
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The cantilever wing uses

- A) external struts or wire bracing.
- B) no external bracing.
- C) the skin to carry most of the load to the wing butt.

## 2. A02A AMA

Which of the following conditions will determine acceptance of wood with mineral streaks?

- A) Careful inspection fails to reveal any decay.
- B) They produce only a small effect on grain direction.
- C) Local irregularities do not exceed limitations specified for spiral and diagonal grain.
- 3. A02A AMA

The I beam wooden spar is routed to

- A) increase strength.
- B) obtain uniform strength.
- C) reduce weight.

## 4. A03A AMA

Glue deterioration in wood aircraft structure is indicated

- A) when a joint has separated and the glue surface shows only the imprint of the wood with no wood fibers clinging to the glue.
- B) when a joint has separated and the glue surface shows pieces of wood and/or wood fibers clinging to the glue.
- C) by any joint separation.

## 5. A03A AMA

Laminated wood is sometimes used in the construction of highly stressed aircraft components. This wood can be identified by its

,	construction.  andard plywood construction.  grain construction.	
In cases of elong  A) it is permissible  B) the spar may lead	A01A gated boltholes in a wood spar or cracks le to ream the hole, plug with hardwood be reinforced by using hardwood reinfor of spar should be spliced in or the span	l, and redrill. rcing plates.
The strength of a	NO1A  It well designed and properly prepared well designed fibers.  It plates.	AMA wood splice joint is provided by the
(1) Machine sew (2) A plain lapped	d seam is never permissible. bove statements, d No. 2 are true. rue.	AMA  be of the folded fell or French fell types.
	n.	AMA ering is based on
of the following p 1. Aluminum foil. 2. Resin impregn	rovide an acceptable protective coating nated cloth tape. ype metal primer.	AMA at come in contact with doped fabric, which j?

B) 1 and 4.		
C) 3 and 4.		
11.	B01A	AMA
-	e (surface tape) is used for	
	event 'ripple formation' in co	-
-		nce under reinforcement tape.
C) To provide	additional wear resistance	over the edges of fabric forming structures.
12.	B01A	AMA
	ing factor(s) for the selection type of aircraft is the	n of the correct weight of textile fabric to be used in
A) maximum v	wing loading.	
B) speed of th	ne aircraft.	
C) speed of the	ne aircraft and the maximur	n wing loading.
13.	C03A	AMA
•	to occur if unhydrated was tes later a finish topcoat, w	h primer is applied to unpainted aluminum and then about hen the humidity is low?
A) Corrosion.		
B) A glossy, b	olush-free finish.	
C) A dull finisl	h due to the topcoat 'sinking	g in' to primer that is still too soft.
14.	C03A	AMA
Before applyir	ng a protective coating to a	ny unpainted clean aluminum, you should
A) wipe the su	urface with avgas or kerose	ne.
B) remove an	y conversion coating film.	
C) avoid toucl	hing the surface with bare h	nands.
15.	C01A	AMA
If registration	numbers are to be applied ce required for the registrat	to an aircraft with a letter height of 12 inches, what is the ion mark N1683C?
Note:		
2/3 x height =	character width.	
1/6 x height =	width for 1.	
1/4 x 2/3 heig	ht = spacing.	
1/6 x height =	stroke or line width.	
A) 52 inches.		

B) 48 inches.		
C) 57 inches.		
16.	C02A	AMA
Which type of coatir application?	ng typically includes phosphoric	acid as one of its components at the time of
A) Wash primer.		
B) Epoxy primer.		
C) Zinc chromate pr	rimer.	
17.	C02A	AMA
Aluminum-pigment i	in dope is used primarily to	
A) provide a silver c		
, .	moisture from the fabric.	
C) reflect ultraviolet		
,		
18.	C04A	AMA
	raft finishes may be caused by a	adverse humidity, drafts, or sudden changes in
temperature?		
A) Orange peel.		
B) Blushing.		
C) Pinholes.		
19.	D07A	AMA
•	compute a bend allowance problal axis of the bend can be	lem and bend allowance tables are not
	ne actual length of the required r	naterial for the bend.
B) found by adding	approximately one half of the sto	ock thickness to the bend radius.
C) found by subtrac	ting the stock thickness from the	bend radius.
20.	D07A	AMA
The sharpest bend t called the	that can be placed in a piece of i	metal without critically weakening the part is
A) bend allowance.		
B) minimum radius o	of bend.	
C) maximum radius	of bend.	
21.	D07A	AMA
A piece of sheet me	tal is bent to a certain radius. Th	ne curvature of the bend is referred to as the

<ul><li>A) bend allowance.</li><li>B) neutral line.</li><li>C) bend radius.</li></ul>		
22.	D07A	AMA
The purpose of a jogo	gle is to	
A) allow clearance fo	r a sheet or an extrusion.	
B) increase obstruction	on for a sheet or an extrusion.	
C) decrease the weig	ht of the part and still retain the neces	sary strength.
23.	D07A	AMA
(Refer to Airframe fig Setback = .252	ure 6.) Determine the dimensions of A	, B, and C in the flat layout.
Bend allowance = .34	15	
A) A = .748; B = 2.25	2; C = 2.004.	
B) $A = .748$ ; $B = 1.49$	6; C = 1.248.	
C) A = 1.252; B = 2.5	04; C = 1.752.	
24.	D07A	AMA
The sight line on a sh marked	neet metal flat layout to be bent in a co	rnice or box brake is measured and
A) one-half radius fro	m either bend tangent line.	
B) one radius from ei	ther bend tangent line.	
C) one radius from th	e bend tangent line that is placed und	er the brake.
25.	D07A	AMA
	ng layout with a single bend, allow for	stretching by
A) adding the setback	•	
<ul><li>B) subtracting the set</li><li>C) subtracting the set</li></ul>	_	
c) subtracting the set	back from botti legs.	
26.	D07A	AMA
You can distinguish b	etween aluminum and aluminum alloy	by
A) filing the metal.	•	•
B) testing with an ace	etic acid solution.	
C) testing with a 10 p	ercent solution of caustic soda.	
27.	D07A	AMA

(Refer to Airframe figure 4.) The length of flat A is	
A) 3.750 inches.	
B) 3.875 inches.	
C) 3.937 inches.	
28. D07A	AMA
(Refer to Airframe figure 5.) What is the flat layout dimensio	n?
A) 7.0 inches.	
B) 6.8 inches.	
C) 6.6 inches.	
29. D02A	AMA
When inspecting a composite panel using the ring test/tappi	ing method, a dull thud may indicate
A) less than full strength curing of the matrix.	
B) separation of the laminates.	
C) an area of too much matrix between fiber layers.	
30. D02A	AMA
One of the best ways to assure that a properly prepared bat to	tch of matrix resin has been achieved is
A) perform a chemical composition analysis.	
B) have mixed enough for a test sample.	
C) test the viscosity of the resin immediately after mixing.	
31. D02A	AMA
When repairing puncture type damage of a metal faced lam the doubler should be tapered to	inated honeycomb panel, the edges of
A) two times the thickness of the metal.	
B) 100 times the thickness of the metal.	
C) whatever is desired for a neat, clean appearance.	
32. D02A	AMA
Sandwich panels made of metal honeycomb construction are type of construction	re used on modern aircraft because this
A) is lighter than single sheet skin of the same strength and	is more corrosion resistant.
B) may be repaired by gluing replacement skin to the inner of	core material with thermoplastic resin.
C) has a high strength to weight ratio.	
33. D02A	AMA

Which of these me water?	thods may be used to in	spect fiberglass/honeycomb structures for entrapped
1. Acoustic emission	on monitoring.	
2. X-ray.		
3. Backlighting.		
A) 1 and 2.		
B) 1 and 3.		
C) 2 and 3.		
34.	D04A	AMA
What is the most c	ommon method of ceme	nting transparent plastics?
A) Heat method.		
B) Soak method.		
C) Bevel method.		
35.	D04A	AMA
If no scratches are surfaces should be	-	plastic enclosure materials have been cleaned, their
A) polished with ru	bbing compound applied	with a damp cloth.
B) buffed with a cle	ean, soft, dry cloth.	
C) covered with a t	thin coat of wax.	
36.	D04A	AMA
Cabin upholstery n	naterials installed in curr	ent standard category airplanes must
A) be fireproof.		
B) be at least flame	e resistant.	
C) meet the require	ements prescribed in Pa	t 43.
37.	D04A	AMA
	nsparent plastic enclosud self-locking nuts, the n	res which are retained by bolts extending through the uts should be
A) tightened to a fire	rm fit, plus one full turn.	
B) tightened to a fire	rm fit, then backed off or	e full turn.
C) tightened to a fi	rm fit.	
38.	D05A	AMA
(Refer to Airframe	figure 2.) Select the pref	erred drawing for proper countersinking.
A) All are acceptab	ole.	

on to
when
heet
ng the

A) centers of rivets in	•	
•	t rivets in the same row.	
C) heads of rivets in t	ne same row.	
45.	D05A	AMA
(Refer to Airframe figumade by a 100° coun	ure 1.) Which of the rivets shown will a tersink?	accurately fit the conical depression
A) 1.		
B) 2.		
C) 3.		
46.	D05A	AMA
What should be the in	ncluded angle of a twist drill for soft me	etals?
A) 118°.		
B) 90°.		
C) 65°.		
47.	D05A	AMA
Shallow scratches in s	sheet metal may be repaired by	
A) burnishing.		
B) buffing.		
C) stop drilling.		
48.	D05A	AMA
	ice is to be used to repair a section of is used, the minimum allowable overla	
A) 1/2 inch.		
B) 3/4 inch.		
C) 13/16 inch.		
49.	D05A	AMA
What is the minimum	edge distance for aircraft rivets?	
A) Two times the dian	neter of the rivet shank.	
B) Two times the dian	neter of the rivet head.	
C) Three times the dia	ameter of the rivet shank.	
50.	D05A	AMA
What is the minimum	spacing for a single row of aircraft rive	ets?

A) Two times the	diameter of the rivet s	shank.
B) Three times th	ne length of the rivet sl	nank.
C) Three times th	ne diameter of the rive	t shank.
51. A potted compou A) 4 inches in dia B) 2 inches in dia C) 1 inch in diam	ameter. ameter.	AMA mb can usually be made on damages less than
1. applying exter 2. room tempera	nal heat. ture exposure. yst or curing agent to t	AMA ng is generally accomplished by the resin.
53. A category of pla A) thermoplastic. B) thermocure. C) thermoset.		AMA pable of softening or flowing when reheated is described as a
A) a 60 percent r B) the orientation	D03A I stiffness of a properly matrix to 40 percent fib n of the plies to the loa he fibers to transfer str	d direction.
55. What is the mate during call A) Bleeder. B) Breather. C) Release.	-	AMA he vacuum bag pressure system to absorb excess resin

56.	D03A	AMA
When making repairs The final cleaning sho	<u> </u>	be repaired is essential for a good bond.
A) MEK (methyl ethyl	ketone).	
B) soap, water, and a	scrub brush.	
C) a thixotropic agent	i.	
57.	D03A	AMA
Fiberglass laminate d	lamage not exceeding the first layer	or ply can be repaired by
A) filling with a putty of	consisting of a compatible resin and	clean, short glass fibers.
B) sanding the damag	ged area until aerodynamic smoothn	ess is obtained.
C) trimming the rough	n edges and sealing with paint.	
58.	D03A	AMA
Fiberglass laminate d	lamage that extends completely thro	ugh one facing and into the core
A) cannot be repaired	d.	
B) requires the replace	cement of the damaged core and faci	ng.
C) can be repaired by	using a typical metal facing patch.	
59.	D03A	AMA
	g, when added to wet resins, provide	
fastener holes in com	•	onengan ter and repair of damaged
1. Microballoons.		
2. Flox.		
3. Chopped fibers.		
A) 2 and 3.		
B) 1 and 3.		
C) 1, 2, and 3.		
60.	D03A	AMA
Which of the following	g are generally characteristic of carbo	on/graphite fiber composites?
1. Flexibility.		
2. Stiffness.		
3. High compressive	strength.	
4. Corrosive effect in		
<b>= A</b> 1 100	contact with aluminum.	
5. Ability to conduct e		

B) 2, 3, and 4. C) 1, 3, and 5.		
61.	D06A	AMA
Which rivets should A) MS20425D-4-3. B) MS20470AD-4-4 C) MS20455DD-5-3		s of .032-inch aluminum?
62.	D06A	AMA
Which of the following A) Rivet diameter. B) Rivet length. C) Type of material		hen determining minimum rivet spacing?
63.	D06A	AMA
A) low strength char B) high alloy conten		
A) 1/8 inch in diame B) 1/8 inch in diame	D06A an MS20430AD-4-8 rivet are eter and 1/4 inch long. eter and 1/2 inch long. neter and 8/32 inch long.	AMA
•		AMA ces of 0.040-inch aluminum riveted together. All gth of the rivets to be used will be
A) a shank length of	D06A dicates a countersunk rivet w f 5/16 inch (excluding head). f 5/32 inch (excluding head).	AMA hich has

C) an overall	length of 5/16 inch.	
67.	D06A	AMA
	rivets in the D and DD series removal from refrigeration	s that are not driven within the prescribed time after heat
A) must be re	heat treated before use.	
B) must be dis	scarded.	
C) may be ret	curned to refrigeration and us	sed later without reheat treatment.
68.	D06A	AMA
	•	two pieces of 0.0625-inch aluminum riveted together. All the length of the rivets to be used will be
A) 5/32 inch.		
B) 3/16 inch.		
C) 5/16 inch.		
69.	D06A	AMA
Which rivet is	used for riveting nickel stee	l alloys?
A) 2024 alum	inum.	
B) Mild steel.		
C) Monel.		
70.	D06A	AMA
A DD rivet is I	heat treated before use to	
A) harden and	d increase strength.	
B) relieve inte	ernal stresses.	
C) soften to fa	acilitate riveting.	
71.	D06A	AMA
The length of equal to	a rivet to be used to join a s	sheet of .032-inch and .064-inch aluminum alloy should be
A) two times t	the rivet diameter plus .064	nch.
B) one and or	ne half times the rivet diame	ter plus .096 inch.
C) three times	s the rivet diameter plus .090	3 inch.
72.	D06A	AMA
Joggles in rer	noved rivet shanks would in	dicate partial
A) bearing fail	lure.	
B) torsion faile	ure.	

C) shear failu	re.	
73.	D06A	AMA
_	rivet to be chosen when ma 4-inch aluminum sheet, drill	aking a structural repair that involves the joining of 0.032 ed with a No. 30 drill, is
74.	D01A	AMA
A main different application) is		kbolt tension and shear fasteners (other than their
A) number of	locking collar grooves.	
B) shape of the		
C) method of	installation.	
75.	D01A	AMA
The general r	rule for finding the proper riv	et diameter is
A) three times	s the thickness of the materi	als to be joined.
,	the rivet length.	
C) three times	s the thickness of the thicke	st sheet.
76.	D01A	AMA
One of the ma	ain advantages of Hi-Lok typ	pe fasteners over earlier generations is that
A) they can b	e removed and reused agai	n.
•	•	vides a more secure, tighter fit.
C) they can b	e installed with ordinary har	id tools.
77.	D01A	AMA
The Dzus turn measured in	nlock fastener consists of a	stud, grommet, and receptacle. The stud diameter is
A) tenths of a	n inch.	
B) hundredth		
C) sixteenths	of an inch.	
78.	D01A	AMA
Hole filling fast primarily because		470 rivets) should not be used in composite structures
A) possibility	of causing delamination.	

B) increased possil	bility of fretting corrosion i	in the fastener.
C) difficulty in formi	ing a proper shop head.	
79.	D01A	AMA
The Dzus turnlock in measured in	fastener consists of a stu-	d, grommet, and receptacle. The stud length is
A) hundredths of a	n inch.	
B) tenths of an inch	٦.	
C) sixteenths of an	inch.	
80.	E03A	AMA
		a steel tube longeron dented at a cluster?
A) Welded split sle		
B) Welded outer sle		
C) Welded patch pl		
-,		
81.	E02A	AMA
The oxyacetylene f	flame for silver soldering s	should be
A) oxidizing.		
B) neutral.		
C) carburizing.		
82.	E04A	AMA
	ed or soldered joints is	
A) not permitted.		
B) permissible for r	mild steel.	
, ·	most metals or alloys that	are not heat treated.
-,,,		
83.	E04A	AMA
A resurfaced solde	ring iron cannot be used	effectively until after the working face has been
A) fluxed.		
B) polished.		
C) tinned.		
84.	E04A	AMA
	urizing flame be avoided	
A) It removes the c	-	<del>u</del>
΄ Β) It hardens the sι		

C) A cold weld will re	esult.	
A) be neutral and so B) be slightly oxidizir		
86.	E05A	AMA
because it		above a certain thickness of aluminum
	tal in alignment during welding.	
•	al or penetration of oxides on the meta	
C) aids in getting full	penetration of the metal and prevents	s local distortion.
87.	E05A	AMA
Where should the flu	x be applied when oxyacetylene weld	ing aluminum?
A) Painted only on th	ne surface to be welded.	
,	rface to be welded and applied to the	welding rod.
C) Applied only to the	e welding rod.	
88.	E05A	AMA
Why are aluminum p	lates 1/4 inch or more thick usually pr	eheated before welding?
A) Reduces internal	stresses and assures more complete	penetration.
B) Reduces welding	time.	
C) Prevents corrosio	n and ensures proper distribution of flo	UX.
89.	E05A	AMA
	oidly when alloys or metals are hot. It is	
A) solvent.		
B) filler.		
C) flux.		
90.	E05A	AMA
Oxygen and acetyler	ne cylinders are made of	
A) seamless aluminu	•	
B) steel.		

C) bronze.		
•	he tensile strength.	AMA
•	material brittle. resses caused by forming	•
A) the penetra B) the penetra	ation should be 100 percen	AMA ected for penetration, ercent of the thickness of the base metal. at of the thickness of the base metal. at the form of a very high bead.
A) Magnesium B) Filler rod sł	E01A ent best describes magnes n can be welded to other m nould be nickel steel. nould be the same compos	netals.
94. The purpose of A) directional of B) longitudina C) lateral stab	l stability.	AMA de
swaging has t A) Measure th B) Use a term	aken place when installing to finished length of the tertional gauge to check the diasurface of the swaged por	AMA s normally used to ensure that the correct amount of swaged-type terminals on aircraft control cable? The minal barrel and compare with the beginning length. The ameter of the swaged portion of the terminal. The swaged portion of the terminal. The swaged portion of the terminal is the swaged portion of the terminal.
terminal, the r A) the full rate	F04A ns issued by the swaging tesultant swaged terminal solutions of the full rated strength of	

C) 70 percent	of the full rated strength of the	cable.
97.	F04A	AMA
A) Corrosion ro B) Anodized a		ompasses, cotter pins are made of what material?
98.	F05A	AMA
the left aileron A) up and the B) down and th	will move elevator will move down. he elevator will move up.	rigged flight controls is moved forward and to the right,
C) down and ti	he elevator will move down.	
99.	F05A	AMA
A) checking fo B) the behavio	or of the trailing edge when the the control surface from its le	hroughout the control surface. e surface is suspended from its hinge points. eading edge in the streamline position and checking
100.	F05A	AMA
Fairleads shou A) 12°. B) 8°. C) 3°.	uld never deflect the alignment	of a cable more than
101.	F05A	AMA
With which sys A) Trim. B) Aileron. C) Elevator.	stem is differential control asso	ociated?
102.	F05A	AMA
lf control cable A) worn attach	es are adjusted properly and th	ne control surfaces tend to vibrate, the probable cause is

C) excessive cable ten	sion.	
103.	F05A	AMA
Stability about the axis A) longitudinal stability. B) lateral stability. C) directional stability.	·	to the line of flight is referred to as
104.	F05A	AMA
the elevator trim tab sy  A) Downward regardles  B) Upward regardless of	stem will cause the ss of elevator position of elevator position	
105.	F05A	AMA
control, side to side mo A) each aileron to have B) each aileron to have	e a greater up trave greater down trave	em of an airplane equipped with differential-type aileron rol stick will cause  I (from the streamlined position) than down travel.  el (from the streamlined position) than up travel.  er number of degrees (from full up to full down) than the
106.	F05A	AMA
The universal propeller A) propeller track. B) aspect ratio of a win C) degrees of flap trave	g.	used to measure
107.	F05A	AMA
A tension regulator in the A) increase the cable to B) provide a means of C) retain a set tension.	ension in cold weat changing cable ten	
108.	F05A	AMA
Differential control on a	an aileron system m	neans that
A) the down travel is m	ore than the up tra	vel.
B) the up travel is more	than the down tra	vel.

C) one aileron on own wash wash		han the aileron on the opposite wing to adjust for
109.	F05A	AMA
If the control stick of the right aileron wil		gged flight controls is moved rearward and to the left
A) down and the el	levator will move down.	
, ·	ator will move down.	
C) down and the el	levator will move up.	
110.	F05A	AMA
What is the smalle	st size cable that may be us	sed in aircraft primary control systems?
A) 1/4 inch.		
B) 5/16 inch.		
C) 1/8 inch.		
111.	F03A	AMA
The correct dihedra	al angle can be determined	by
A) measuring the a	angular setting of each wing	at the rear spar with a bubble protractor.
B) placing a straight position.	ntedge and bubble protracto	or across the spars while the airplane is in flying
C) using a dihedral	l board and bubble level alo	ng the front spar of each wing.
112.	F03A	AMA
Where would you f aircraft?	ind precise information to p	erform a symmetry alignment check for a particular
A) Aircraft Specific	ation or Type Certificate Da	ta Sheet.
B) Manufacturer's	service bulletins.	
C) Aircraft service	or maintenance manual.	
113.	F03A	AMA
Where is fuselage	station No. 137 located?	
A) 137 centimeters	aft of the nose or fixed refe	erence line.
B) 137 inches aft o	of the zero or fixed reference	e line.
C) Aft of the engine	9.	
114.	F06A	AMA
Why is it generally	necessary to jack an aircra	ft indoors for weighing?
A) So aircraft may	be placed in a level position	1.

B) So that air currents do r C) So weighing scales ma	y be calibrated to 0 pounds.	
115.	F02A	AMA
If the vertical fin of a single parallel to	e engine, propeller driven airplane i	s rigged properly, it will generally be
A) the longitudinal axis but	t not the vertical axis.	
B) the vertical axis but not	the longitudinal axis.	
C) both the longitudinal an	d vertical axes.	
116.	F02A	AMA
As the angle of attack of a	n airfoil increases, the center of pre	essure will
A) move toward the trailing	g edge.	
B) remain stationary becau angle of attack.	use both lift and drag components i	increase proportionally to increased
C) move toward the leadin	g edge.	
117.	F02A	AMA
An airplane which has god	od longitudinal stability should have	a minimum tendency to
A) roll.		
B) pitch.		
C) yaw.		
118.	F02A	AMA
An airplane's center of lift i	is usually located aft of its center of	f gravity
A) so that the airplane will	have a tail heavy tendency.	
B) so that the airplane will	have a nose heavy tendency.	
C) to improve stability abo	ut the longitudinal axis.	
119.	F02A	AMA
The primary purpose of sta	all strips is to	
A) provide added lift at slo	w speeds.	
B) stall the inboard portion	of the wings first.	
C) provide added lift at hig	h angles of attack.	
120.	F02A	AMA
The chord of a wing is mea	asured from	
A) wingtip to wingtip.		

B) wing root to the wingtip. C) leading edge to trailing e	edge.	
121. What physical factors are in A) Thickness and chord. B) Span and chord. C) Dihedral and angle of att	F02A evolved in the aspect ratio of airpla	AMA ne wings?
122. An airplane that has a tendemotion has A) poor longitudinal stability B) good lateral stability. C) poor lateral stability.	F02A ency to gradually increase a pitchir	AMA ng moment that has been set into
123. The angle of incidence of an All All All All All All All All All Al	e wings in flight. between the relative wind and the	AMA chord of the wing.
avoided, the aircraft should  A) obliquely into the wind.	e it makes no difference if the wind	
A) around or about the long B) around or about the later C) around or about the long 126.	F02A ong its lateral axis (roll) is also more itudinal axis controlled by the eleva- ral axis controlled by the ailerons. itudinal axis controlled by the ailer F01A free wheeling unit in a helicopter de	ator. rons. AMA
	•	s below the equivalent of rotor RPM.

B) It releases the rotor brake for starting	
C) It relieves bending stress on the rotor	blades during starting.
127. F01A	AMA
Which statement is correct concerning to	orque effect on helicopters?
A) Torque direction is the same as rotor	blade rotation.
B) As horsepower decreases, torque inc	creases.
C) Torque direction is the opposite of rot	tor blade rotation.
128. F01A	AMA
One purpose of the freewheeling unit red is to	quired between the engine and the helicopter transmission
A) automatically disengage the rotor from	m the engine in case of an engine failure.
B) disconnect the rotor from the engine	to relieve the starter load.
C) permit practice of autorotation landing	gs.
129. F01A	AMA
If a single rotor helicopter is in forward h	orizontal flight, the angle of attack of the advancing blade is
A) more than the retreating blade.	
B) equal to the retreating blade.	
C) less than the retreating blade.	
130. F01A	AMA
A helicopter in forward flight, cruise conf	iguration, changes direction by
A) varying the pitch of the main rotor bla	des.
B) changing rotor RPM.	
C) tilting the main rotor disk in the desire	ed direction.
131. F01A	AMA
The auxiliary (tail) rotor of a helicopter post of the following?	ermits the pilot to compensate for and/or accomplish which
A) Attitude and airspeed.	
B) Lateral and yaw position.	
C) Torque and directional control.	
132. G01A	AMA
Large airplanes and turbine-powered mu Regulation Part 91, General Operating a	ultiengine airplanes operated under Federal Aviation and Flight Rules, must be inspected

A) in accordance with Subpart E.	an inspection program	authorized under Federal Aviation Regulation Part 91,
	n a continuous airworthi on Regulation Part 91, S	ness maintenance program (camp program) authorized Subpart E.
C) in accordance with Section 91.409(d).	n the progressive inspec	ction requirements of Federal Aviation Regulation
133.	G01A	AMA
	•	raft that is found to be unairworthy after an annual repair (assuming approved data is used to accomplish
A) An appropriately rately rate for return to service.	ated mechanic may acc	complish the repair, and an IA may approve the aircraft
B) An appropriately rately rately rater.	ated mechanic or repair	station may repair the defect and approve the aircraft
C) Only the person who service, after the major		al inspection may approve the aircraft for return to
134.	G01A	AMA
Which statement abo	ut Airworthiness Directi	ves (AD's) is true?
A) AD's are information manufacturer.	on alert bulletins issued	by the airframe, powerplant, or component
B) Compliance with a	n AD is not mandatory	unless the aircraft affected is for hire.
<ul><li>C) Compliance with a records.</li></ul>	n applicable AD is man	datory and must be recorded in the maintenance
135.	K01A	AMA
Aircraft tire pressure s	should be checked	
A) using only a push only a push only at least once a week		ving 1-pound increments.
C) as soon as possibl		
136.	K01A	AMA
What should be checl A) Air pressure.	ked when a shock strut	bottoms during a landing?
B) Packing seals for c	correct installation.	
C) Fluid level.		
137.	K01A	AMA
Overinflated aircraft ti	ires may cause damage	to the

A) brake linings. B) wheel hub. C) wheel flange.		
A) Provides an internal B) Straightens the nose	·	AMA shock strut?
A) reduce brake presso B) relieve excessive flu	K01A e used in brake systems primarily ture and maintain static pressure.  uid and ensure a positive release.  e to the brake and increase the volu	
140. The purpose of a relief A) reduce pressure for B) prevent the tire from C) compensate for the	n skidding.	AMA
141. The fusible plugs instal A) indicate tire tread se B) prevent overinflation C) melt at a specified e	).	AMA
A) collapsed and fluid a B) fully extended and f	K01A oil shock strut with MIL-5606 the standard at the filler opening. Iuid added at the filler opening. and fluid added at the filler opening.	
A) ensure rapid applica	K01A used in brake systems primarily to ation and release of the brakes. ure and maintain static pressure.	AMA

C) reduce the pre	essure and release the brakes	rapidly.
144.	K01A	AMA
•	out of tolerance toe in condition or twisted components consist	on of main landing gear wheels determined not to be
A) shimming the	axle in the oleo trunnion.	
B) inserting, rem the scissor torqu		n of washers or spacers at the center pivotal point of
C) placing shims	or spacers behind the bearing	g of the out of tolerance wheel or wheels.
145.	K01A	AMA
The primary purp	oose for balancing aircraft whe	el assemblies is to
A) prevent heavy	spots and reduce vibration.	
	aircraft weight properly.	
	sive wear and turbulence.	
146.	K01A	AMA
	urce for power brakes is	
A) the main hydr	•	
B) the power bra		
C) a master cylin		
147.	K01A	AMA
	in a brake master cylinder unit	
A) slow release o	•	
,	lowly creep down while pedal p	oressure is applied
C) fading brakes		stoccare to applica.
e, raamig brance	•	
148.	K01A	AMA
The purpose of a	a sequence valve in a hydraulio	c retractable landing gear system is to
A) prevent heavy	landing gear from falling too r	apidly upon extension.
	ans of disconnecting the norma	al source of hydraulic power and connecting the
C) ensure operat	tion of the landing gear and ge	ar doors in the proper order.
149.	K01A	AMA
The purpose of a	an orifice check valve is to	
	re to a sensitive component.	
	one direction and allow free fl	ow in the other

C) relieve pressure i	n one direction and prevent	flow in the other direction.
150.	K01A	AMA
A special bolt in a la foot-pounds are requ A) 36.8. B) 38. C) 36.6.		ires a torque value of 440 inch-pounds. How many
151.	K01A	AMA
marked with	, ,	using MIL-H-5606 (mineral base) fluid will be
A) a blue stripe or do		
B) one or more white		
C) a white and yello	w sinpe.	
152.	K01A	AMA
Which statement is t type brake assembli		Ift equipped with hydraulically operated multiple disk
A) There are no min compensating cylind		rance checks required due to the use of self
B) Do not set parking	g brake when brakes are ho	t.
C) No parking brake	provisions are possible for	this type of brake assembly.
153.	K01A	AMA
What device in a hydno demands are on A)  A) Pressure relief van  B) Shuttle valve.	the system?	nt delivery pump allows circulation of the fluid when
C) Pressure regulate	or.	
154.	K01A	AMA
How long should you	u wait after a flight before ch	ecking tire pressure?
A) At least 2 hours (	3 hours in hot weather).	
B) At least 3 hours (	4 hours in hot weather).	
C) At least 4 hours (	5 hours in hot weather).	
155.	K01A	AMA
Lockout deboosters	are primarily pressure reduc	cing valves that

A) allow full debooste pressure chamber.	er piston travel without fluid	from the high pressure side entering the low
B) cannot allow full d pressure chamber.	lebooster piston travel witho	out fluid from the high pressure side entering the low
C) must be bled sepa	arately after brake bleeding	has been completed.
156.	K01A	AMA
		ing gear would most likely be a
A) shunt field series		
B) split field shunt wo C) split field series w		
e) opiit noid conce ii		
157.	K01A	AMA
		provide a warning in the cockpit when the throttle is
,	r is not down and locked.	
	ar is down and locked.	
C) retarded and geal	r is down and locked.	
158.	K01A	AMA
		aircraft tire is an indication of
A) incorrect camber.		
B) excessive toe out.		
C) overinflation.		
159.	K01A	AMA
When an empty shoostrut completely at le		re should be taken to extend and compress the
A) thoroughly lubrica	te the piston rod.	
B) force out any exce	ess fluid.	
C) ensure proper pa	cking ring seating and remo	val of air bubbles.
160.	K01A	AMA
In shock struts, chev	ron seals are used to	
A) absorb bottoming		
B) prevent oil from e		
C) serve as a bearing	g surface.	
161.	K01A	AMA
How can it be detern	nined that all air has been p	urged from a master cylinder brake system?

<ul><li>A) By operating a deflection.</li></ul>	a hydraulic unit and watching the sys	tem pressure gauge for smooth, full scale
B) By noting whe	ether the brake is firm or spongy.	
C) By noting the	amount of fluid return to the master of	cylinder upon brake release.
162.	K01A	AMA
In brake service	work, the term 'bleeding brakes' is th	e process of
A) withdrawing a	ir only from the system.	
		of removing air that has entered the system.
C) replacing sma	all amounts of fluid in reservoir.	
163.	K01A	AMA
		master cylinder will have on a brake system?
•	ill operate normally.	
•	will be filled by reverse flow.	
C) The restriction	n will cause slow release of the brake	·S.
164.	K01A	AMA
What would be t	he effect if the piston return spring br	oke in a brake master cylinder?
A) The brakes w	ould become spongy.	
B) The brake tra	vel would become excessive.	
C) The brakes w	ould drag.	
165.	K01A	AMA
	e right brake on an aircraft is spongy The probable cause is	when the brake pedal is depressed in a
A) the hydraulic	master cylinder piston is sticking.	
B) air in the brak	e hydraulic system.	
C) the hydraulic	master cylinder piston return spring is	s weak.
166.	K01A	AMA
The metering pir	ns in oleo shock struts serve to	
A) lock the struts	s in the DOWN position.	
B) retard the flow	v of oil as the struts are compressed.	
C) meter the pro	per amount of air in the struts.	
167.	K01A	AMA
	wheel manufacturers often recomme the wheel from the axle?	nd that the tires on split rim wheels be deflated

A) To relieve the strain o	n the wheel retaining	nut and axle threads.
B) As a safety precautior or weakened.	n in case the bolts tha	at hold the wheel halves together have been damaged
C) To remove the static I	oad imposed upon th	e wheel bearings by the inflated tire.
168.	K01A	AMA
adaptable to mechanical	operation?	mechanically or hydraulically. Which type is not
A) Single disk spot type.		
B) Single servo type.		
C) Expander tube type.		
169.	K01A	AMA
Exposure to and/or stora	ige near which of the	following is considered harmful to aircraft tires?
1. Low humidity.		
2. Fuel.		
3. Oil.		
4. Ozone.		
5. Helium.		
6. Electrical equipment.		
7. Hydraulic fluid.		
8. Solvents.		
A) 2, 3, 4, 5, 6, 7, 8.		
B) 1, 2, 3, 5, 7, 8.		
C) 2, 3, 4, 6, 7, 8.		
170.	L02A	AMA
Two types of hydraulic flu A) mineral base, and pho		sed in civil aircraft are
B) mixed mineral base a	nd phosphate ester b	ase.
C) petroleum base and n	nixed mineral base.	
171.	L02A	AMA
(1) Materials which are S polyurethane and epoxy	-	resistant include most common aircraft metals and
(2) Skydrol hydraulic fluid	d is compatible with n	ylon and natural fibers.
Regarding the above sta	tements,	
A) neither No. 1 nor No.	2 is true.	

3) both No. 1 and No. 2 are C) only No. 1 is true.	true.	
172.  How can the proper hydraul  A) Refer to the aircraft parts  B) Consult the aircraft Type  C) Consult the aircraft manu	Certificate Data Sheet.	AMA be determined?
173. What is used to flush a syste  A) Methyl ethyl ketone or ke  B) Naphtha or varsol.  C) Lacquer thinner or trichlo		AMA 5606 hydraulic fluid?
174. Where can information be o aircraft materials? A) Manufacturer's technical B) Aircraft manufacturer's sp C) AC 43.13-1A.		AMA fire resistant hydraulic fluid with
175. The internal resistance of a A) volatility. B) viscosity. C) acidity.	L02A fluid which tends to prevent it fror	AMA n flowing is called
manufacturer's maintenance		e affixed to the reservoir or unit.
177.	L02A	AMA

_	•	by atmospheric humidity if left unprotected?
1. MIL-H-5606 hydraulic		
2. Skydrol hydraulic fluid	I <b>.</b>	
3. None of the above.		
A) 1 and 2.		
B) 3.		
C) 2.		
178.	L03A	AMA
Which must be done bef pressure regulator?	ore adjusting the reli	ef valve of a main hydraulic system incorporating a
A) Eliminate the action o	f the unloading valve	€.
B) Adjust all other syster	n relief valves which	have a lower pressure setting.
C) Manually unseat all sy	ystem check valves	to allow unrestricted flow in both directions.
179.	L03A	AMA
The unit which causes o	ne hydraulic operatio	on to follow another in a definite order is called a
A) selector valve.		
B) sequence valve.		
C) shuttle valve.		
180.	L03A	AMA
Severe kickback of the e indicate which of the following		hand pump handle during the normal intake stroke will
A) The hand pump inlet	check valve is stickir	ng open.
B) The main system relie	ef valve is set too hig	ıh.
C) The hand pump outle	t check valve is stick	ing open.
181.	L03A	AMA
The main system pressurvalve should be adjusted		mple hydraulic system equipped with a power control
A) with the power contro	I valve held in the CI	_OSED position.
B) while one or more act	uating units are in o	peration.
C) with the power contro	I valve in the OPEN	position.
182.	L03A	AMA
•		lled and air chamber charged, the main system Iraulic pressure reading until

A) at least one selector accumulator.	valve has been a	ctuated to allow fluid to flow into the fluid side of the
B) the air pressure has C) the fluid side of the a	-	•
183. Which seals are used v A) Polyester. B) Butyl rubber.	L03A vith petroleum ba	AMA se hydraulic fluids?
C) Buna-N.		
184.	L03A	AMA
•	t closed if the fluid a	evice which is designed to remain open to allow a normal flow increases above an established rate. This device is
185.	L03A	AMA
How is the air in a hydr	aulic accumulator	prevented from entering the fluid system?
<ul><li>A) By forcing the oil/air leaving the accumulato</li></ul>	•	centrifugal separating chamber that prevents the air from
B) By physically separa separator.	ating the air cham	per from the oil chamber with a flexible or movable
C) By including a valve	that automatically	closes when the fluid level lowers to a preset amount.
186.	L03A	AMA
The primary function of	the flap overload	valve is to
A) prevent the flaps from	m being lowered	at airspeeds which would impose excessive structural loads.
	craft will not beco	oposite sides of the aircraft centerline to extend and retract ne aerodynamically unbalanced to the extent that it
C) boost normal systen relatively large flap area	-	laps in order to overcome the air loads acting on the
187.	L03A	AMA
If it is necessary to adjusequence, if any, shoul		re regulating valves in a hydraulic system, what particular
A) Units most distant from	om the hydraulic	oump should be adjusted first.

B) Units with the higl	hest pressure settings are	e adjusted first.	
C) Units are indepen	ndent of each other, and the	herefore, no particular sequence is necessary.	
188.	L03A	AMA	
Unloading valves are	e used with many engine	driven hydraulic pumps to	
A) dampen out press	sure surges.		
B) relieve the pump	pressure.		
C) relieve system pro	essure.		
189.	L03A	AMA	
Which is true regardiinstalled?	ing the ground check of a	flap operating mechanism which has just been	
•	d to operate the mechanied out of the system.	sm increases with successive operations, it indicate	es:
•	d to operate the mechanied out of the system.	sm decreases with successive operations, it indicate	es
C) All hydraulic lines connections.	and components should	be checked for leaks by applying soapy water to all	
190.	L03A	AMA	
flaps cannot be lowe		ground runup of an aircraft indicates that the wing ulic system, but can be lowered by using the ely cause?	
A) The flap selector	valve has a severe interna	al leak.	
B) The pressure acc	umulator is not supplying	pressure to the system.	
C) The fluid level in t	the reservoir is low.		
191.	L03A	AMA	
	•	mp, it is found that the handle cannot be moved in t t likely cause is an incorrectly installed	the
A) hand pump inport	check valve.		
B) inport/outport orifi	ce check valve.		
C) hand pump outpo	ort check valve.		
192.	L03A	AMA	
are connected to the A) different amounts			;

C) equal amounts of fo	orce and will move at t	he same rate of speed.
193.	L03A	AMA
	ng units are required in high rates of fluid flow	some aircraft hydraulic systems because of
194.	L03A	AMA
Although dents in the	heel of a bend are not	permissible, they are acceptable in the remainder of a hat percent of the tube diameter?
195.	L03A	AMA
If hydraulic fluid is rele of	ased when the air val	ve core of the accumulator is depressed, it is evidence
A) excessive accumula	•	
B) a leaking check val		
C) a ruptured diaphraç	ym or leaking seals.	
196.	L03A	AMA
Hydraulic system them A) lower pressure than B) higher pressure than C) lower pressure than	n the system relief valv an the system relief val	ve.
197.	L03A	AMA
How would the air pres but the system still has	_	cumulator be determined if the engine is inoperative,
A) Read it directly from	n the main system pre	ssure gauge with all actuators inoperative.
B) Build up system pre attached to the air side	_	ency pump and then read the pressure on a gauge
C) Operate a hydraulic goes toward zero.	c unit slowly and note	the pressure at which a rapid pressure drop begins as it
198.	L03A	AMA
What is the main purp	ose of a pressurized re	eservoir in a hydraulic system?

B) Prevent hydra	ollapse at altitude. ulic pump cavitation. ulic fluid from foaming.	
199.	L03A	AMA
A) cleaned and re B) discarded at re	ering elements constructed of eused.  egular intervals and replaced for use in certificated aircraft.	with new filtering elements.
200.	L03A	AMA
Hydraulic system  1. Dampen press  2. Supplement the  3. Store power fo	accumulators serve which o ure surges. e system pump when demar	f the following functions?  Indicate the following functions?  Indicate the following functions?  Indicate the following functions?
201.	L03A	AMA
Quick disconnect	couplings in hydraulic syste	ms provide a means of
A) easily replacin	g hydraulic lines in areas wh	ere leaks are common.
B) quickly conned contaminates ent		aulic lines and eliminate the possibility of
C) quickly connect the system.	cting and disconnecting hydr	aulic lines without loss of fluid or entrance of air into
202.	L03A	AMA
A hydraulic pump	is a constant-displacement	type if it
A) produces an u	nregulated constant pressure	e.
B) produces a co	ntinuous positive pressure.	
C) delivers a unifo	orm rate of fluid flow.	
203.	L03A	AMA
pump is stopped,	no hydraulic pressure is ava	s running, the pressure is normal. However, when the illable. This is an indication of a
<ul><li>A) leaking selector</li></ul>	valve.	

B) low accumulato	r fluid preload.	
C) leaking accumu	ılator air valve.	
204.	L03A	AMA
-	em that has a reservoir press the air pressure between the	surized with turbine engine compressor bleed air, e engine and reservoir?
A) Relief valve.		
B) Air bleed relief v	valve.	
C) Air pressure reg	gulator.	
205.	L03A	AMA
Teflon hose that hat temperature should		et from being exposed to high pressure or
A) not be straighte	ned or bent further.	
B) not be reinstalle	ed once removed.	
C) be immediately	replaced.	
206.	L03A	AMA
A hydraulic motor	converts fluid pressure to	
A) linear motion.		
B) rotary motion.		
C) angular motion.		
207.	L01A	AMA
What is one advan	stage of piston type hydraulic	motors over electric motors?
A) They are consid	derably quieter in operation.	
B) There is no fire	hazard if the motor is stalled	
C) They work satis	sfactorily over a wider temper	ature range.
208.	L01A	AMA
The installation of	a new metal hydraulic line sh	ould be made with
A) a straight tube t	o withstand the shocks and v	vibration to which it will be subjected.
B) a straight tube t leakage.	o permit proper alignment of	the fitting and thereby reduce fluid loss through
C) enough bends t vibration.	to allow the tube to expand a	nd contract with temperature changes and to absorb
209.	L01A	AMA
Which characterist	tics apply to aircraft hydraulic	systems?

1. Minimum mainten	ance requirements.	
<ol> <li>Lightweight.</li> <li>About 80 percent</li> <li>Simple to inspect.</li> </ol>		ercent loss due to fluid friction).
A) 1, 2, 3, 4.		
B) 1, 3, 4.		
C) 1, 2, 4.		
210.	L01A	AMA
The purpose of the p	oressure regulator in a hyd	raulic system is to
,		a predetermined range and to unload the pump. ating cylinders within the system.
, •		hydraulic lines under excessive pressure.
211.	L01A	AMA
	or valve is one of the most f fluid into and out of a con	commonly used in hydraulic systems to provide for nected actuating unit?
A) Four port, closed	center valve.	
B) Three port, four w	vay valve.	
C) Two port, open ce	enter valve.	
212.	L01A	AMA
Pneumatic systems	utilize	
A) return lines.		
B) relief valves.		
C) diluter valves.		
213.	L01A	AMA
compressor, also red	-	tes an engine driven multistage reciprocating
A) an oil separator.		
B) a surge chamber.		
C) a moisture separa	ator.	
214.	L01A	AMA
What type of packing containing Skydrol?	gs should be used in hydra	aulic components to be installed in a system
A) AN packings mad	de of natural rubber.	

<ul><li>B) Packing materia</li><li>C) AN packings ma</li></ul>	als made for ester base fluids ade of neoprene.	3.
215.	L01A	AMA
		t directs pressurized fluid to one end of an actuating to the reservoir from the other end.
216.	L01A	AMA
		craft hydraulic units, the most commonly used type of
217.	L01A	AMA
A) check valve. B) orifice check va C) selector valve.		used to direct the flow of fluid is the
218.	L01A	AMA
(Refer to Airframe A) 1. B) 2. C) 3.	figure 11.) Which fitting is ar	AN flared tube fitting?
219.	M01A	AMA
What test is used to A) Pressure test w B) Pressure test w C) Pressure test w	ith nitrogen.	y of an oxygen cylinder?
220.	M01A	AMA
•	• .	f liquid refrigerant is introduced into the low side of a is too high or the outside air temperature is too low?

B) Condenser.		
C) Evaporator.		
224	N404 A	0 B 4 0
221. When charging a vapor cy out of a vacuum. What is i		AMA the low pressure gauge fails to come
A) Blockage in the system		
B) The expansion valve fa	iled to close.	
C) The compressor is not	engaging.	
222.	M01A	AMA
Frost or ice buildup on a v	apor cycle cooling system evaporate	or would most likely be caused by
A) the mixing valve stickin	g closed.	
B) moisture in the evapora		
C) inadequate airflow thro	ugn the evaporator.	
223.	M01A	AMA
take a freon charge?	or cycle cooling system would most I	ikely be at fault if a system would not
A) Expansion valve.		
B) Condenser.		
C) Receiver dryer.		
224.	M01A	AMA
•	aircraft in flight is maintained at the s	selected altitude by
A) controlling the air inflow      Infloting door cools and		
C) controlling the rate at w	recirculating conditioned cabin air.	
c) controlling the rate at w	Then all leaves the capill.	
225.	M01A	AMA
What is ventilating air use	d for on a combustion heater?	
A) Provides combustion a	ir for ground blower.	
B) Carries heat to the plac	ces where needed.	
C) Provides air required to	support the flame.	
226.	M01A	AMA
	d weight high pressure oxygen cylind	
A) Every 5 years.		•

B) Every 4 years.		
C) Every 3 years.		
227.	M01A	AMA
•	apor cycle cooling system, if the me temperature, what does the	ne two lines connected to the expansion valve are is indicate?
A) The system is	functioning normally.	
B) The expansion	valve is not metering freon pr	operly.
C) The compress	or is pumping too much refrige	erant.
228.	M01A	AMA
When Refrigerant	t 12 is passed over an open fla	ame, it
A) changes to me	ethane gas.	
B) is broken dowr	n into its basic chemical eleme	ents.
C) changes to phe	osgene gas.	
229.	M01A	AMA
The purpose of a	subcooler in a vapor cycle co	oling system is to
A) augment the c	ooling capacity during periods	of peak demand.
B) aid in quick co	oling a hot aircraft interior.	
C) cool the freon	to prevent premature vaporiza	ition.
230.	M01A	AMA
	reon air conditioning system, i on for the slow rate discharge?	t is important to release the charge at a slow rate.
A) Prevent the lar	ge amount of freon from conta	aminating the surrounding atmosphere.
B) Prevent excess	sive loss of refrigerant oil.	
C) Prevent conde	ensation from forming and conf	taminating the system.
231.	M01A	AMA
What is the condi	tion of the refrigerant as it leav	ves the condenser of a vapor cycle cooling system?
A) Low pressure I	liquid.	
B) High pressure	liquid.	
C) High pressure	vapor.	
232.	M01A	AMA
The evacuation o	f a vapor-cycle cooling system	removes any water that may be present by
A) drawing out the	e liquid.	

B) raising the bo	iling point of the water and di	rawing out the vapor.
C) lowering the b	poiling point of the water and	drawing out the vapor.
233.	M01A	AMA
The point at whic from a liquid to a		por cycle cooling system absorbs heat and changes
A) condenser.		
B) evaporator.		
C) expansion val	lve.	
234.	M01A	AMA
What componenairplane altitude?		prevents the cabin altitude from becoming higher than
A) Cabin rate of	descent control.	
B) Negative pres	sure relief valve.	
C) Positive press	sure relief valve.	
235.	M01A	AMA
The function of the	he evaporator in a freon cool	ing system is to
A) liquefy freon i	n the line between the compr	essor and the condenser.
B) lower the tem	perature of the cabin air.	
C) transfer heat f	from the freon gas to ambien	t air.
236.	M01A	AMA
On some cabin p	pressurization systems, press	surization on the ground is restricted by the
A) cabin pressur	e regulator.	
B) negative pres	sure-relief valve.	
C) main landing	gear operated switch.	
237.	M01A	AMA
How is the cabin	pressure of a pressurized ai	rcraft usually controlled?
	·	the pressurization pump to turn on or off as required.
		all the pressure in excess of the amount for which it is
C) By a pressure	e sensitive valve that controls	the output pressure of the pressurization pump.
238.	M01A	AMA
		por cycle cooling system gives up heat and changes

A) condenser.		
B) evaporator.		
C) expansion valv	e.	
239.	M01A	AMA
The main cause o	f contamination in gaseous	oxygen systems is
A) moisture.		
B) dust and other	airborne particulates.	
C) other atmosphe	eric gases.	
240.	M01A	AMA
The altitude contro	oller maintains cabin altitude	by modulation of the
A) safety and outf	low valves.	
B) safety valve.		
C) outflow valve.		
241.	M01A	AMA
	ed a good practice concerning jacket around the engine ext	ng the inspection of heating and exhaust systems of naust as a heat source?
A) Supplement ph	ysical inspections with perio	dic operational carbon monoxide detection tests.
· ·	tem components should be article inspection method.	removed periodically, and their condition determined
C) All exhaust sys period.	stem components should be	removed and replaced at each 100-hour inspection
242.	M01A	AMA
What type of oxyg	en system uses the rebreath	ner bag-type mask?
A) Diluter demand	d.	
B) Continuous flow	W.	
C) Demand.		
243.	M01A	AMA
A pressurization c	controller uses	
A) bleed air press	ure, outside air temperature,	and cabin rate of climb.
B) barometric pres	ssure, cabin altitude, and cal	oin rate of change.
C) cabin rate of cl	imb, bleed air volume, and c	abin pressure.
244.	M01A	AMA
Which best descri	bes cabin differential pressu	re?

A) Difference bet	ween cabin flight altitude pre	ssure and Mean Sea Level pressure.
B) Difference bet	ween the ambient and intern	al air pressure.
C) Difference bet	ween cabin pressure control	er setting and actual cabin pressure.
245.	M01A	AMA
•	e figure 13.) Determine what in a freon refrigeration syster	unit is located immediately downstream of the n.
A) Condenser.		
B) Compressor.		
C) Evaporator co	ils.	
246.	M01A	AMA
The operation of	an aircraft combustion heate	r is usually controlled by a thermostat circuit which
A) alternately turr	ns the fuel on and off, a proce	ess known as cycling.
B) meters the am BTU output.	ount of fuel continuously ent	ering the heater and therefore regulates the heater's
C) regulates the	voltage applied to the heater'	s ignition transformer.
247.	M01A	AMA
One purpose of a	i jet pump in a pressurization	and air conditioning system is to
A) produce a high	n pressure for operation of th	e outflow valve.
B) provide for aug	gmentation of airflow in some	e areas of the aircraft.
C) assist in the ci	rculation of freon.	
248.	M01A	AMA
When servicing a	n air conditioning system tha	t has lost all of its freon, it is necessary to
A) check oil and a	add as necessary, evacuate	the system, relieve vacuum, and add freon.
B) check oil and a	add as necessary, evacuate	the system, and add freon.
C) check oil and a	add as necessary, and add fi	reon.
249.	M01A	AMA
The function of a	n expansion valve in a freon	cooling system is to act as a metering device and to
A) reduce the pre	essure of the gaseous freon.	
B) increase the p	ressure of the liquid freon.	
C) reduce the pre	essure of the liquid freon.	
250.	M01A	AMA
The cabin pressu	rization modes of operation a	are

•	ential, and maximum differentia	al.
•	oressurized, and isobaric.	
C) ambient, unpre	essurized, and isobaric.	
251.	M02A	AMA
What controls the	amount of oxygen delivered to	a mask in a continuous flow oxygen system?
<ul><li>A) Calibrated orific</li></ul>	ce.	
B) Pressure reduc	cing valve.	
C) Pilot's regulato	r.	
252.	M02A	AMA
		d a leak, the lines and fittings should be
A) removed and re	•	
	g a special oxygen system dye	
C) bubble tested v	with a special soap solution ma	anufactured specifically for this purpose.
253.	M02A	AMA
Oxygen systems i	n unpressurized aircraft are ge	enerally of the
A) continuous flow	v and pressure demand types.	
B) pressure dema	nd type only.	
C) portable bottle	type only.	
254.	M02A	AMA
Before a high pres	ssure oxygen cylinder is servic	ed, it must be the correct type and have been
A) hydrostatically	tested within the proper time in	nterval.
B) approved by th	e National Transportation Safe	ety Board.
C) inspected by a	certificated airframe mechanic	<b>&gt;.</b>
255.	M02A	AMA
The purpose of the	e airflow metering aneroid ass	embly found in oxygen diluter demand regulators is
A) regulate airflow positions.	in relation to oxygen flow who	en operating in emergency or diluter demand
B) regulate airflow	in relation to cabin altitude wl	nen in diluter demand position.
C) automatically p	out the regulator in emergency	position if the demand valve diaphragm ruptures.
256.	M02A	AMA
(1) Oxygen used i	n aircraft systems is at least 9	9.5 percent pure and is practically water free.

	·	ercent pure and is hospital quality.
Regarding the abov		
A) only No. 1 is true B) both No. 1 and N		
C) neither No. 1 no		
257.	M02A	AMA
In a gaseous oxyge skin?	en system, which of the foll	owing are vented to blow out plugs in the fuselage
A) Pressure relief v	alves.	
B) Filler shutoff val	ves.	
C) Pressure reduce	er valves.	
258.	M02A	AMA
High pressure cylin	ders containing oxygen for	r aviation use can be identified by their
A) green color and	the words 'BREATHING C	XYGEN' stenciled in 1-inch white letters.
B) yellow color and	the words 'AVIATOR'S BF	REATHING OXYGEN' stenciled in 1-inch white letters.
C) green color and	the words 'AVIATOR'S BR	REATHING OXYGEN' stenciled in 1-inch white letters.
259.	M02A	AMA
259. If a high pressure o the		AMA talled in an airplane, it must meet the specifications of
If a high pressure o the		talled in an airplane, it must meet the specifications of
If a high pressure o the	exygen cylinder is to be insturer or the cylinder manuf	talled in an airplane, it must meet the specifications of
If a high pressure o the A) aircraft manufac B) Department of T	exygen cylinder is to be insturer or the cylinder manuforansportation.	talled in an airplane, it must meet the specifications of
If a high pressure o the A) aircraft manufac B) Department of T C) National Transp	exygen cylinder is to be insturer or the cylinder manuforansportation.	talled in an airplane, it must meet the specifications of acturer.
If a high pressure on the  A) aircraft manufact  B) Department of T  C) National Transport  260.	exygen cylinder is to be insturer or the cylinder manuforansportation.  Ortation Safety Board or the N01A	talled in an airplane, it must meet the specifications of acturer.  e Standards of Compressed Gas Cylinders.
If a high pressure on the A) aircraft manufact B) Department of T C) National Transpo 260. A radar altimeter in	exygen cylinder is to be institurer or the cylinder manuforansportation.  ortation Safety Board or the N01A dicates	talled in an airplane, it must meet the specifications of acturer.  e Standards of Compressed Gas Cylinders.
If a high pressure of the A) aircraft manufact B) Department of T C) National Transpo 260. A radar altimeter in A) flight level (press	exygen cylinder is to be institurer or the cylinder manuforansportation.  ortation Safety Board or the N01A dicates sure) altitude.	talled in an airplane, it must meet the specifications of acturer.  e Standards of Compressed Gas Cylinders.
If a high pressure o the A) aircraft manufac B) Department of T	exygen cylinder is to be institurer or the cylinder manuforansportation.  ortation Safety Board or the N01A dicates sure) altitude.	talled in an airplane, it must meet the specifications of facturer.  e Standards of Compressed Gas Cylinders.
If a high pressure of the  A) aircraft manufact B) Department of T C) National Transport  260. A radar altimeter inc A) flight level (press B) altitude above se C) altitude above ge	exygen cylinder is to be institurer or the cylinder manuforansportation.  ortation Safety Board or the N01A dicates sure) altitude.	talled in an airplane, it must meet the specifications of facturer.  e Standards of Compressed Gas Cylinders.
If a high pressure of the A) aircraft manufact B) Department of T C) National Transpo 260. A radar altimeter in A) flight level (press B) altitude above go C) altitude above go	exygen cylinder is to be institurer or the cylinder manuforansportation.  ortation Safety Board or the N01A dicates sure) altitude. ea level. round level.	talled in an airplane, it must meet the specifications of facturer.  e Standards of Compressed Gas Cylinders.  AMA
If a high pressure of the A) aircraft manufact B) Department of T C) National Transport 260. A radar altimeter inc A) flight level (press B) altitude above se C) altitude above ge 261. A radar altimeter de A) transmitting a sig	turer or the cylinder manufuransportation. ortation Safety Board or the N01A dicates sure) altitude. ea level. round level. N01A etermines altitude by gnal and receiving back a level.	talled in an airplane, it must meet the specifications of facturer.  e Standards of Compressed Gas Cylinders.  AMA  AMA  reflected signal.
If a high pressure of the A) aircraft manufact B) Department of T C) National Transport 260. A radar altimeter inc A) flight level (press B) altitude above se C) altitude above ge 261. A radar altimeter de A) transmitting a sig B) receiving signals	exygen cylinder is to be instituted or the cylinder manuforansportation.  ortation Safety Board or the N01A dicates sure) altitude. ea level. round level.  N01A N01A etermines altitude by	talled in an airplane, it must meet the specifications of facturer.  e Standards of Compressed Gas Cylinders.  AMA  AMA  reflected signal.

262.	N01A	AMA
A Bourdon tube i	nstrument may be used to inc	licate
1. pressure.		
2. temperature.		
3. position.		
A) 1 and 2.		
B) 1.		
C) 2 and 3.		
263.	N01A	AMA
The operating me	echanism of most hydraulic pr	essure gauges is
A) a Bourdon tub	e.	
B) an airtight diap	hragm.	
C) an evacuated attached.	bellows filled with an inert ga	s to which suitable arms, levers, and gears are
264.	N01A	AMA
(1) Aircraft instrur	ments are color-coded to dire	ct attention to operational ranges and limitations.
Regulations but a	re standardized by aircraft m	t specified by Title 14 of the Code of Federal anufacturers.
Regarding the ab		
A) only No. 1 is tr		
B) only No. 2 is tr		
C) both No. 1 and	d No. 2 are true.	
265.	N01A	AMA
	result if the instrument static per during cruising flight?	ressure line becomes disconnected inside a
A) The altimeter a	and airspeed indicator will bot	h read low.
B) The altimeter a	and airspeed indicator will bot	h read high.
C) The altimeter	will read low and the airspeed	l indicator will read high.
266.	N01A	AMA
Magnetic compas	ss bowls are filled with a liquid	d to
A) retard precess	ion of the float.	
B) reduce deviati	on errors.	
C) dampen the o	scillation of the float.	

267.	N01A	AMA
The maximum altitude system integrity check		an unpressurized aircraft instrument static pressure
A) 50 feet in 1 minute	·.	
B) 200 feet in 1 minut	ie.	
C) 100 feet in 1 minut	te.	
268.	N01A	AMA
When an aircraft altim	neter is set at 29.92 inch	nes Hg on the ground, the altimeter will read
A) pressure altitude.		
B) density altitude.		
C) field elevation.		
269.	N01A	AMA
The function of a sym	nbol generator (SG) in a	n EFIS is to
A) display alphanume	eric data and representa	tions of aircraft instruments.
B) allow the pilot to se	elect the appropriate sys	stem configuration for the current flight situation.
<ul><li>C) receive and proces appropriate display.</li></ul>	ss input signals from air	craft and engine sensors and send the data to the
270.	N01A	AMA
Data transmitted betw	veen components in an	EFIS are converted into
A) digital signals.		
B) analog signals.		
C) carrier wave signa	ls.	
271.	N01A	AMA
The requirements for are contained in	testing and inspection of	of instrument static systems required by Section 91.411
A) Type Certificate Da	ata Sheets.	
B) AC 43.13-1A.		
C) Part 43, appendix	E.	
272.	N01A	AMA
Who is authorized to	repair an aircraft instrun	nent?
1. A certified mechani	ic with an airframe rating	g.
2. A certificated repair	rman with an airframe ra	ating.
3. A certificated repair	r station approved for th	at class instrument.

<ol> <li>A certificated a</li> </ol>	irframe repair station.	
A) 1, 2, 3, and 4.		
B) 3 and 4.		
C) 3.		
273.	N01A	AMA
Fuel flow transmit	ters are designed to transmit	data
A) mechanically.	-	
B) electrically.		
C) utilizing fluid po	ower.	
274.	N01A	AMA
•	an angle of attack indicating s he airstream flows in a directi	system is based on detection of differential pressure on
A) not parallel to t	he true angle of attack of the	aircraft.
B) parallel to the a	angle of attack of the aircraft.	
C) parallel to the I	ongitudinal axis of the aircraf	t.
275.	N01A	AMA
What does a recipoperating?	procating engine manifold pre	ssure gauge indicate when the engine is not
A) Zero pressure.		
•	•	ure and the atmospheric pressure.
C) The existing at	mospheric pressure.	
276.	N02A	AMA
	_	their respective panels depends on the
A) instrument mai		
B) design of the ir		
C) design of the ir	nstrument panel.	
277.	N02A	AMA
	t panels are generally shock	mounted to absorb
A) all vibration.		
	high amplitude shocks.	
C) high frequency	, high amplitude shocks.	
278.		AMA

What marking color i	is used to indicate if a cov	er glass has slipped?
A) Red.		
B) White.		
C) Yellow.		
279.	N02A	AMA
The green arc on an	aircraft temperature gaug	e indicates
A) the instrument is i	not calibrated.	
B) the desirable temp	perature range.	
C) a low, unsafe tem	perature range.	
280.	N02A	AMA
Which procedure sho	ould you use if you find a	vacuum operated instrument glass loose?
A) Mark the case and	d glass with a slippage ma	ark.
B) Replace the glass	<b>3.</b>	
C) Install another ins	strument.	
281.	N02A	AMA
An aircraft instrumen	nt panel is electrically bond	ded to the aircraft structure to
A) act as a restraint	strap.	
B) provide current re	turn paths.	
C) aid in the panel in	stallation.	
282.	N02A	AMA
Where may a persor engine instrument?	n look for the information n	necessary to determine the required markings on an
1. Engine manufactu	rer's specifications.	
2. Aircraft flight manu	ual.	
3. Instrument manufa	acturer's specifications.	
4. Aircraft maintenan	ice manual.	
A) 2 or 4.		
B) 1 or 4.		
C) 2 or 3.		
283.	N02A	AMA
A certificated mecha	nic may perform	
A) minor repairs to in	nstruments.	
B) 100-hour inspection	ons of instruments.	

C) instrument overhaul.		
284. The lubber line on a dire A) represent the nose of B) align the instrument of C) represent the wings	of the aircraft. glass in the case.	AMA
285.	N02A	AMA
requirements of Section tester?  1. Vertical speed indica 2. Cabin altimeter. 3. Altimeter. 4. Cabin rate-of-change 5. Airspeed indicator.	n 91.411, what aircraft tor.	ure system is leak checked to comply with the instrument may be used in lieu of a pitot-static system
A) 1 or 5.		
B) 2 or 4. C) 3.		
<ul><li>286.</li><li>When performing the st</li><li>A) static pressure.</li><li>B) positive pressure.</li><li>C) negative pressure.</li></ul>	N02A atic system leakage ch	AMA neck required by Section 91.411, the technician utilizes
287.	O02A	AMA
	ft, what electronic devi	ce typically monitors flight parameters and performs
288.	O02A	AMA
In general, the purpose	of an aircraft transpon	der is to
A) continually transmit h	neading, speed, and ra	ate of climb/decent etc. information to ATC.
B) monitor aircraft spee	d, heading, altitude, ar	nd attitude whenever the autopilot system is engaged.

C) receive an interrogation	on signal from a ground	station and automatically send a reply back.
289.	O02A	AMA
Static dischargers help eatmosphere at  A) low current levels.  B) high voltage level.  C) high current levels.	eliminate radio interferen	ce by dissipating static electricity into the
290.	O02A	AMA
An aircraft antenna insta A) to the airframe. B) to the engine. C) to the radio rack.	llation must be grounded	t
291.	O02A	AMA
When must the radio sta A) When the aircraft is o B) When the aircraft is re C) When the aircraft is c	perated outside the U.S. eturned to service.	d in an aircraft equipped with a two-way radio?
292.	O02A	AMA
Part of the ADF system	used on aircraft includes	
A) RMI indicator antenna	ā.	
B) marker beacon anten		
C) sense and loop anten	ınas.	
293.	O02A	AMA
The preferred location of	f an ELT is	
A) where it is readily acc	essible to the pilot or a r	member of the flightcrew while the aircraft is in flight
B) as far aft as possible.		
C) as far aft as possible,	but forward of the vertic	al fin.
294.	O02A	AMA
How may the battery rep	lacement date be verifie	ed for an emergency locator transmitter (ELT)?
<ul><li>A) By removing the batte the useful life remains.</li></ul>	eries and testing them ur	nder a measured load to determine if 50 percent of
B) By observing the batt	ery replacement date ma	arked on the outside of the transmitter.

C) By activating the	transmitter and measuri	ng the signal strength.	
295.	O01A	AMA	
In an autopilot, which A) Displacement signs) Course signal. C) Followup signal.		t signal to the ailerons?	
296.	O01A	AMA	
In which control ele A) Command. B) Sensing. C) Input.	ment of an autopilot syste	em is an attitude indicator?	
297.	O01A	AMA	
What component of A) Servo. B) Controller. C) Gyro.	an autopilot system app	ies torque to the control surfaces of an aircraft?	
298.	O01A	AMA	
Which channel of an A) Elevator. B) Aileron. C) Rudder.	n autopilot detects chang	es in pitch attitude of an aircraft?	
299.	O01A	AMA	
What component is A) Servo. B) Gyro. C) Controller.	the sensing device in an	electromechanical autopilot system?	
300.	O01A	AMA	
•	checking an autopilot sy he autopilot should be er	stem on the ground, after the aircraft's main powe gaged	r has
	ros come up to speed and	I the amplifier warms up.	
B) whenever the op C) for only a few mi			
Up for only a few fill	nuits ai a liiit.		

301.	O01A	AMA
Dutch roll, a combinatio counteracted with	n yawing and rolling os	scillation that affects many sweptwing aircraft, is
A) a flight director syste	m.	
B) an aileron damper sy	vstem.	
C) a yaw damper syster	n.	
302.	O03A	AMA
When an antenna is ins	talled, it should be fast	ened
A) to the primary structu	ire at the approximate	intersection of the three aircraft axes.
B) with a reinforcing dou	ubler on each side of th	ne aircraft skin.
C) so that loads impose	d are transmitted to the	e aircraft structure.
303.	O03A	AMA
Doublers are used when	n antennas are installe	d to
A) eliminate antenna vib	oration.	
B) prevent oil canning o	f the skin.	
C) reinstate the structur	al strength of the aircra	aft skin.
304.	O03A	AMA
What characteristics of the evaluated?	the installation of a rigi	d antenna on a vertical stabilizer should be
A) Polarization and impe	edance.	
B) Impedance and inter	ference.	
C) Flutter and vibration.		
305.	O03A	AMA
(Refer to Airframe figure of .125 square feet insta	,	oximate drag load on an antenna with a frontal area
A) 2.069 pounds.		a speca e. <b>11</b> 0
B) 2.073 pounds.		
C) 2.080 pounds.		
0) 2.000 pourids.		
306.	O03A	AMA
A DME antenna should	be located in a positio	n on the aircraft that will
A) not be blanked by the	e wing when the aircra	ft is banked.
B) permit interruptions in	n DME operation.	
C) eliminate the possibil	lity of the DME locking	on a station.

307.	O03A	AMA
When installing A) null position.	a DME antenna, it should be al	igned with the
B) angle of incid	dence.	
C) centerline or	n the airplane.	
308.	O03A	AMA
(Refer to Airfrar	me figure 16.) Which of the ante	nnas shown is a typical glideslope antenna?
A) 2.		
B) 3.		
C) 4.		
309.	O03A	AMA
		equired when installing radio equipment under a seat?
,	n the seat unoccupied.	accives adequate cooling and damage protection
•		eceives adequate cooling and damage protection.  to maximum downward seat spring deflection.
310.	P01A	AMA
Fuel is moved of	overboard in most fuel jettison s	ystems by
A) boost pumps	S.	
B) gravity.		
C) gravity and e	engine driven fuel pumps.	
311.	P01A	AMA
Which of the fol	llowing is employed to maintain	lateral stability when jettisoning fuel?
A) Two separat	e independent systems.	
B) Crossfeed sy	ystem.	
C) Two intercor	nnected systems.	
312.	P01A	AMA
The primary pu	rpose of an aircraft's fuel jettisor	າ system is to quickly achieve a
A) lower landing	-	
B) balanced fue		
C) reduced fire	hazard.	
313.	P03A	AMA

Aircraft pressure fueling s	ystems instructional procedures are	normally placarded on the
A) fuel control panel acce	ss door.	
B) lower wing surface adja	acent to the access door.	
C) aircraft ground connec	tion point.	
314.	P03A	AMA
Which of the following pre	ecautions is most important during re	efueling operations?
-	ources must be disconnected from the	•
B) Fuel to be used must b		
C) All electrical switches r	• • •	
315.	P07A	AMA
The primary purpose of a	fuel tank sump is to provide a	
A) positive system of mair	ntaining the design minimum fuel su	pply for safe operation.
B) place where water and	dirt accumulations in the tank can o	collect and be drained.
C) reserve supply of fuel t	to enable the aircraft to land safely in	n the event of fuel exhaustion.
316.	P07A	AMA
How may the antiknock cl	naracteristics of a fuel be improved?	
A) By adding a knock inhi	bitor.	
B) By adding a knock enh	ancer.	
C) By adding a fungicide a	agent.	
317.	P07A	AMA
What is the maximum vap	oor pressure allowable for an aircraft	fuel?
A) 7 PSI.		
B) 5 PSI.		
C) 3 PSI.		
318.	P07A	AMA
The vapor pressure of avi	ation gasoline is	
A) lower than the vapor p	ressure of automotive gasoline.	
B) higher than the vapor p C) approximately 20 PSI a	oressure of automotive gasoline. at 100 °F.	
319.	P07A	AMA
(1) If aviation gasoline vapincreased fuel flow.	porizes too readily, fuel lines may be	ecome filled with vapor and cause

(2) A measure of a Regarding the abo A) only No. 2 is tru B) both No. 1 and C) neither No. 1 no	ve statements, e. No. 2 are true.	apor lock is obtained from the Reid vapor pressure test.
320.	P07A	AMA
` '		stem, a pressure refueling receptacle and control panel or all fuel tanks of an aircraft.
(2) Because of the aircraft.	fuel tank area, there are	more advantages to a pressure fueling system in light
Regarding the abo	ve statements,	
A) only No. 1 is tru	e.	
B) only No. 2 is tru	e.	
C) both No. 1 and	No. 2 are true.	
321.	P07A	AMA
	_	e tank which is known to be uncontaminated with dirt or ank sumps and system strainers
	ed except for the strainer 100-hour or annual insp	check before the first flight of the day and the fuel tank ections.
B) are still necessa	ary due to the possibility o	of contamination from other sources.
,	reduced since contamina odern aircraft fuel system	tion from other sources is relatively unlikely and of little is.
322.	P07A	AMA
What type of fuel b	ooster pump requires a p	oressure relief valve?
A) Concentric.		
B) Sliding vane.		
C) Centrifugal.		
323.	P07A	AMA
	23, what minimum require reciprocating engine-pow	ed markings must be placed at or near each appropriate rered airplanes?
A) The word 'Avga	s' and the minimum fuel (	grade.
B) The word 'Fuel'	and usable fuel capacity	
C) The word 'Avga	s' and the total fuel capa	city.
324.	P07A	AMA

Why are centrifugal typ A) Because they are por B) To supply fuel under C) To permit cooling air	ositive displacement r pressure to engine	e driven pumps.	
325.	P07A	AMA	
Flapper valves are use	d in fuel tanks to		
A) reduce pressure.			
B) prevent a negative p	ressure.		
C) act as check valves.			
326.	P07A	AMA	
Fuel boost pumps are of	operated		
A) to provide a positive	flow of fuel to the	engine.	
B) primarily for fuel tran	nsfer.		
C) automatically from for	uel pressure.		
327.	P07A	AMA	
What minimum required utility category aircraft?	_	e placed on or near each appropriate fuel filler cover on	
A) The word 'Avgas' an	d the minimum fue	I grade, and the total fuel tank capacity.	
B) The word 'Avgas' an tank capacity.	d the minimum fue	I grade or designation for the engines, and the usable fuel	
C) The word 'Avgas' an	nd the minimum fue	l grade .	
328.	P07A	AMA	
What precautions must be observed if a gravity feed fuel system is permitted to supply fuel to an engine from more than one tank at a time?			
A) The tank airspaces i	must be interconne	cted.	
B) The fuel outlet ports	of each tank must	have the same cross sectional area.	
C) Each tank must have empty.	e a valve in its outle	et that automatically shuts off the line when the tank is	
329.	P05A	AMA	
What must each fuel quel remaining is equal	•	calibrated to read during level flight when the quantity of el supply?	
A) The total unusable f	uel quantity.		
B) Both the total unusa	ble fuel quantity an	d the unusable fuel quantity in each tank.	
C) Zero.			

330.	P05A	AMA	
A) It sends an election A) It senses the to	se of a float operated tran ctric signal to the fuel qua stal amount of fuel density ielectric qualities of fuel an	•	
A) float operated t B) float resting on	P05A fuel quantity indicating systems ransmitter installed in the the surface of the tank. receiver installed in the tale		а
332. How does temper A) Cold fuel is hea B) Warm fuel is he C) Temperature h	eavier per gallon.	AMA	
333. A capacitance typ A) pounds. B) pounds per hou C) gallons.	P05A e fuel quantity indicating s ur.	AMA system measures fuel in	
A) the indicators a B) only one transr	re calibrated in gallons; th	AMA uel quantity indicating systems is that nerefore, no conversion is necessary. e needed regardless of the number of tanks. ne indicator.	
335. A probe or a serie A) Selsyn. B) Capacitor. C) Synchro.	P05A s of probes is used in wha	AMA at kind of fuel quantity indicating system?	
336.	P05A	AMA	

A drip gauge may be used to the amount of fuel in the B) system leakage with the C) fuel pump diaphragm lea	tank. system shut down. akage.	
337. The probe of a capacitance A) float actuated variable ca B) capacitor with fuel and a C) capacitor with fuel and a	ir acting as one plate.	AMA a
338. Fuel system components m A) drain off static charges. B) prevent stray currents. C) retard galvanic corrosion	P02A ust be bonded and grounded in ord	AMA der to
A) calls for jettisoning of fue B) reduces contamination a	P02A em operation in multiengine aircraft el overboard to correct lateral instat nd/or fire hazards during fueling or intain a balanced fuel load conditio	oility. defueling operations.
340. Which procedure must be for A) Defuel all the tanks at on B) Defuel the inboard wing to Defuel the outboard wing	tanks first.	AMA n sweptback wings?
341. What is the recommended p A) Purge the tank with air. B) Flush the inside of the ta C) Steam clean the tank int		AMA fore welding?
•	P04A aircraft's fuel tank, which proceduk during the entire work period.	AMA re should be avoided?

•		cess to perform rescue operations if required. ration in an air conditioned building.
A) To maintain a B) To decrease f	P04A  bose of a fuel tank vent?  ttmospheric pressure.  fuel vapor pressure.  tank internal air pressure.	AMA
344. Which of the follon A) Welding and r B) Brazing and r C) Riveting and	resealing. esealing.	AMA air of fuel leaks on most integral fuel tanks?
A) It traps any sr B) It provides a d	P04A fuel strainer located at the lower mall amount of water that may be drain for residual fuel. Taps all micro organisms that methods.	·
346.	P04A	AMA
What method wo		al leakage of a fuel valve without removing the valve
	ve in the OFF position, drain the to the strainer bowl.	e strainer bowl, and with boost pump on, watch to
•	ed air pressure on the downstre	and watch for bubbling in the tanks. eam side of the fuel pump and listen for air passing
347.	P04A	AMA
When moving th RPM should	e mixture control on a normally	operating engine into the idle cutoff position, engine
	se before the engine starts to o	lie.
, ,	ase and then drop rapidly. ame until the cutoff is effected, t	hen drop rapidly.
348.	P06A	AMA
(1) A fuel pressu	re relief valve is required on an	aircraft positive displacement fuel pump.

(2) A fuel pressure relie	ef valve is require	d on an aircraft centrifugal fuel boost pump.			
Regarding the above st	tatements,				
A) only No. 1 is true.					
	B) only No. 2 is true.				
C) both No. 1 and No. 2	2 are true.				
349.	P06A	AMA			
(1) A fuel heater can us	se engine bleed a	ir as a source of heat.			
(2) A fuel heater can us	se engine lubricat	ing oil as a source of heat.			
Regarding the above st	tatements,				
A) only No. 1 is true.					
B) both No. 1 and No. 2	2 are true.				
C) neither No. 1 nor No	o. 2 is true.				
350.	P06A	AMA			
(1) The function of a fue	el heater is to pro	tect the engine fuel system from ice formation.			
(2) An aircraft fuel heat	er cannot be use	d to thaw ice in the fuel screen.			
Regarding the above st	tatements,				
A) only No. 1 is true.					
B) only No. 2 is true.					
C) both No. 1 and No. 2	2 are true.				
351.	P06A	AMA			
Which of the following vaircraft fuel system?	would be most us	seful to locate and troubleshoot an internal fuel leak in an			
A) Aircraft structure rep	air manual.				
B) Illustrated parts man	ıual.				
C) A fuel system schen	natic.				
352.	P06A	AMA			
A fuel pressure warning	g switch contacts	close and warning light is turned on when			
A) a measured quantity	of fuel has pass	ed through it.			
B) the fuel flow stops.					
C) the fuel pressure dro	ops below specifi	ed limits.			
353.	P06A	AMA			
	n turbine powere	d aircraft to determine when the condition of the fuel is			
approaching the dange					

A) Fuel pressure warnii	ng.	
B) Fuel pressure gauge	<b>)</b> .	
C) Fuel temperature inc	dicator.	
354.	P06A	AMA
What is the purpose of	flapper type check va	alves in integral fuel tanks?
A) To allow defueling o	f the tanks by suction	٦.
B) To prevent fuel from	flowing away from the	ne boost pumps.
C) To allow the engine	driven pumps to drav	w fuel directly from the tank if the boost pump fails.
355.	P06A	AMA
What unit would be adj	usted to change the	fuel pressure warning limits?
A) Fuel flowmeter bypa	ss valve.	
B) Pressure sensitive m	nechanism.	
C) Fuel pressure relief	valve.	
356.	Q04A	AMA
CSD driven generators	are usually cooled b	у
A) oil spray.		
3) an integral fan.		
C) both ram air and an	integral fan.	
357.	Q04A	AMA
ntegrated drive genera	itors (IDG) employ a	type of high output ac generator that utilizes
A) brushes and slip ring	gs to carry generated	dc exciter current to the rotating field.
B) battery current to ex	cite the field.	
C) a brushless system	to produce current.	
358.	Q04A	AMA
When necessary during	g operation, CSD disc	connect is usually accomplished by
A) a switch in the cockp	oit.	
3) circuit breaker activa	ation.	
C) a shear section in th	e input shaft.	
359.	Q03A	AMA
One advantage of using	g ac electrical power	in aircraft is
A) that ac electrical mo	tors can be reversed	while dc motors cannot.
B) greater ease in step	ping the voltage up o	or down.

C) that the effect power input is re		aximum instantaneous voltage; therefore, less
360.	Q03A	AMA
A) greater and th B) greater and th	n ac transformer secondary that be amperage less than in the pri be amperage greater than in the amperage greater than in the pri	primary.
361.	Q03A	AMA
What is a method A) Flash the field B) Reseat the bro C) Energize the a	ushes.	eld residual magnetism?
362.	Q03A	AMA
B) Use shielded	n independent of the position lig electrical cable to assure fail sa anticollision light to the aircraft p	fe operation.
363.	Q03A	AMA
component and s A) the componer B) either the com	g an electrical circuit, if an ohm some value of resistance is read at has continuity and is open.  apponent or the circuit is shorted at has continuity and is not oper	
364.	Q03A	AMA
How are generat A) Watts at rated B) Amperes at ra C) The impedand	l voltage.	
365.	Q03A	AMA
	enerator are laminated to	
A) reduce flux los	sses.	
B) increase flux of	concentration.	

C) reduce eddy	current losses.	
366.	Q03A	AMA
The inductor type	oe inverter output voltage is con	trolled by the
A) number of po	oles and the speed of the motor	•
B) voltage regu	lator.	
C) dc stator fiel	d current.	
367.	Q03A	AMA
If any one gene	rator in a 24-volt dc system sho	ws low voltage, the most likely cause is
A) an out of adj	ustment voltage regulator.	
B) shorted or gr	ounded wiring.	
C) a defective r	everse current cutout relay.	
368.	Q03A	AMA
A voltage regula	ator controls generator voltage l	by changing the
A) resistance in	the generator output circuit.	
B) current in the	e generator output circuit.	
C) resistance of	f the generator field circuit.	
369.	Q03A	AMA
_	•	supply power for a single load, their controls include s share the load equally. The equalizer circuit
A) increasing th	e output of the low generator to	equal the output of the high generator.
B) decreasing t	he output of the high generator	to equal the output of the low generator.
C) increasing the they are equal.	ne output of the low generator a	nd decreasing the output of the high generator until
370.	Q03A	AMA
The most community the	non method of regulating the vo	Itage output of a compound dc generator is to vary
A) current flowing	ng through the shunt field coils.	
B) total effective	e field strength by changing the	reluctance of the magnetic circuit.
C) resistance of	f the series field circuit.	
371.	Q03A	AMA
•	me figure 19.) Upon completion the red light remained lit. What	of the landing gear extension cycle, the green light is the probable cause?

A) Short in the do	wn limit switch.		
B) Short in the gear safety switch.			
C) Short in the up	limit switch.		
372.	Q03A	AMA	
-		ternators) as a primary source of electrical power charging through the use of	
A) a stepdown tra	nsformer and a rectifier.		
B) an inverter and	l a voltage dropping resisto	r.	
C) a dynamotor w	rith a half wave dc output.		
373.	Q03A	AMA	
Major adjustments on equipment such as regulators, contactors, and inverters are best accomplished outside the airplane on test benches with necessary instruments and equipment. Adjustment procedure should be as outlined by  A) the equipment manufacturer.  B) the FAA.			
C) aircraft technic	al orders.		
374.	Q03A	AMA	
(Refer to Airframe	e figure 18.) Which of the ba	atteries are connected together incorrectly?	
A) 1.			
B) 2.			
C) 3.			
375.	Q02A	AMA	
A) Coaxial cables B) Coaxial cables	ting of coaxial cables differ are routed parallel with str are routed at right angles t are routed as directly as p	to stringers or ribs.	
376.	Q02A	AMA	
A certain switch is indicates the num		e, double throw switch (SPDT). The throw of a switch	
A) circuits each po	ole can complete through tl	ne switch.	
B) terminals at wh	nich current can enter or lea	ave the switch.	
C) places at which time open or close		gle, plunger, etc.) will come to rest and at the same	

377.	Q02A	AMA
What is an impo	ortant factor in selecting aircraf	t fuses?
A) The current e	exceeds a predetermined value	€.
B) The voltage r	ating should be lower than the	e maximum circuit voltage.
C) Capacity mat	tches the needs of the circuit.	
378.	Q02A	AMA
What is the adva	antage of a circuit breaker who	en compared to a fuse?
A) Never needs	replacing.	
B) Always elimir	nates the need of a switch.	
C) Resettable a	nd reusable.	
379.	Q02A	AMA
does not limit th	_	enerator or alternator lead, and the regulator system enerator or alternator can deliver, the ammeter can be ternator rating?
B) 75.		
C) 100.		
,		
380.	Q02A	AMA
Bonding connec	ctions should be tested for	
A) resistance va	alue.	
B) amperage va	lue.	
C) reactance.		
381.	Q02A	AMA
	cables must pass through hole cted from chafing by	s in bulkheads, formers, ribs, firewalls, etc., the wires
A) wrapping with	n electrical tape.	
B) using a suital	ble grommet.	
C) wrapping with	h plastic.	
382.	Q02A	AMA
lf it is necessary should	to use an electrical connecto	r where it may be exposed to moisture, the mechanic
A) coat the conr	nector with grease.	
B) use a special	l moisture proof type.	
C) spray the cor	nnector with varnish or zinc ch	romate.

383.	Q02A	AMA
If a wire is installed given the wire?	d so that it comes in conta	act with some moving parts, what protection should be
A) Wrap with soft v	wire solder into a shield.	
B) Wrap with friction	on tape.	
C) Pass through c	onduit.	
384.	Q02A	AMA
What is the voltage operation?	e drop for a No. 18 coppe	er wire 50 feet long to carry 12.5 amperes, continuous
Use the formula V	D = RLA	
VD = Voltage drop	)	
R = Resistance pe	er ft = .00644	
L = Length of wire		
A = Amperes		
A) 1/2V.		
B) 1V.		
C) 4V.		
385.	Q02A	AMA
A circuit breaker is	s installed in an aircraft el	ectrical system primarily to protect the
A) circuit and shou	uld be located as close to	the source as possible.
B) circuit and shou	uld be located as close to	the unit as possible.
C) electrical unit in	n the circuit and should be	e located as close to the source as possible.
386.	Q02A	AMA
Electric circuits are	e protected from overhea	ting by means of
A) thermocouples.		
B) shunts.		
C) fuses.		
387.	Q02A	AMA
Aircraft fuse capac	city is rated in	
A) volts.		
B) ohms.		
C) amperes.		
388.	Q02A	AMA

endanger a system freque A) guarded switches.	ated only in an emergency or whose ently employ reakers only (no switches).	inadvertent activation could		
C) spring-loaded to off tog	gle or rocker switches.			
389.	Q02A	AMA		
_	aircraft without special enclosing me tallation, simple maintenance, and re	· ·		
A) be limited as to the nur	mber of cables to minimize damage f	rom a single electrical fault.		
B) include at least one shi	elded cable to provide good bonding	of the bundle to the airframe.		
C) be limited to a minimur stresses on the cable insu	n bend radius of five times the bundle lation.	e diameter to avoid excessive		
390.	Q02A	AMA		
Which of the following should be accomplished in the installation of aircraft wiring?  A) Support the bundle to structure and/ or solid fluid lines to prevent chafing damage.  B) Provide adequate slack in the wire bundle to compensate for large changes in temperature.  C) Locate the bundle above flammable fluid lines and securely clamp to structure.				
391.	Q02A	AMA		
9	When using the voltage drop method of checking circuit resistance, the  A) input voltage must be maintained at a constant value.			
, .	maintained at a constant value.			
C) input voltage must be v				
392.	Q02A	AMA		
Aircraft electrical junction	boxes located in a fire zone are usua	ally constructed of		
A) asbestos.				
B) cadmium plated steel.				
C) stainless steel.				
393.	Q02A	AMA		
The primary consideration	ns when selecting electric cable size	are		
A) current carrying capaci	ty and allowable voltage drop.			
B) the voltage and ampera	age of the load it must carry.			
C) the system voltage and	C) the system voltage and cable length.			

394.	Q01A	AMA
What is the color	and orientation of the position	on lights for navigation on civil airplanes?
A) Left side - gree	en, right side - red, rear aft -	white.
B) Left side - red,	right side - green, rear aft -	white.
C) Left side - whit	e, right side - green, rear af	- red.
395.	Q01A	AMA
	ic wire terminals used for m apabilities, are designed pri	ost aircraft applications, in addition to providing good marily
A) to prevent circu	uit failure due to terminal dis	connection.
B) for uncomplica	ted and rapid circuit connec	tion and disconnection.
C) for permanent	connection to the circuit.	
396.	Q01A	AMA
The commutator of	of a generator	
A) changes direct armature.	current produced in the arn	nature into alternating current as it is taken from the
B) changes altern armature.	ating current produced in th	e armature into direct current as it is taken from the
C) reverses the c	urrent in the field coils at the	proper time in order to produce direct current.
397.	Q01A	AMA
In a generator, whof the brushes wit		sparking to the brush guides caused by the movement
A) The brush pigt		
B) Brush spring te	ension.	
C) Undercutting tl	he mica on the commutator.	
398.	Q01A	AMA
A voltage regulate	or controls generator output	by
A) introducing a re	esistance in generator-to-ba	ttery lead in the event of overload.
B) shorting out fie	eld coil in the event of overlo	ad.
C) varying curren	t flow to generator field coil.	
399.	Q01A	AMA
What is the most	accurate type of frequency	neasuring instrument?
	uit chip having a clock circui	_
	ometers using electromagne	
C) Flectromagnet	s using one permanent mad	inet.

400.	Q01A	AMA	
What does a red	ctifier do?		
A) Changes dire	ect current into alternating cur	rent.	
B) Changes alte	rnating current into direct cur	rent.	
C) Reduces volt	age.		
401.	Q01A	AMA	
Which of the foll	owing is not one of the purpo	oses of interpoles in a generator?	
A) Reduce field	-		
•	mature reaction.		
C) Reduce arcin	g at the brushes.		
402.	Q01A	AMA	
One purpose of	a growler test is to determine	the presence of	
 A) an out of rour		·	
B) a broken field	l lead.		
C) a shorted arn	nature.		
400	0044	A B 4 A	
403.	Q01A	AMA	
-	cipal advantage of the series	wound ac motor?	
A) High starting  B) Suitable for a			
C) Low starting	onstant speed use.		
C) LOW Starting	iorque.		
404.	Q01A	AMA	
A series wound	dc electric motor will normall	y require	
A) more current	at high RPM than at low RPI	٧I.	
B) approximately	y the same current throughou	ut its operating range of speed.	
C) more current	at low RPM than at high RPI	VI.	
405.	Q01A	AMA	
	·	the commutator bars of a dc generator undercut	?
-	width of the mica.	tuo commutator sare er a de gemerator amacreat	
•	e the width of the mica.		
	width of the mica.		
, ,			
406.	Q01A	AMA	

A) interpoles.	ed in overcoming the effect of armat	ture reaction is through the use of
B) shaded poles. C) drum wound armatures	in combination with a negatively co	nnected series field.
•		
407.	Q01A	AMA
•	ised in a dc generator are a part of	the
A) armature assembly.     B) field assembly.		
C) brush assembly.		
o) brach accombly.		
408.	Q01A	AMA
	It relay contact points fail to open af rent will flow through the generator	ter the generator output has dropped armature
<ul><li>A) in the normal direction a</li><li>B) and the shunt field oppo</li></ul>	and through the shunt field opposite site the normal direction.	the normal direction.
C) opposite the normal dire	ection and through the shunt field in	the normal direction.
409.	Q01A	AMA
To test generator or motor	armature windings for opens,	
A) place armature in a grov light.	wler and connect a 110V test light o	n adjacent segments; light should
,	ts on commutator with an ohmmete between the armature core segment	•
410.	Q01A	AMA
For general electrical use in A) crimping. B) soldering. C) crimping and soldering.	n aircraft, the acceptable method of	attaching a terminal to a wire is by
411.	Q01A	AMA
How can it be determined i  A) Measure the input voltage  B) The output voltage will b  C) The transformer will get	pe high.	f its turns shorted together?
412.	Q01A	AMA

Which of the follouse for an aircraf	•	nto consideration when determining the wire size to
1. Mechanical str		
2. Allowable pow	•	
3. Ease of install	ation.	
4. Resistance of	current return path through th	e aircraft structure.
5. Permissible vo	oltage drop.	
6. Current carryir	ng capability of the conductor.	
7. Type of load (	continuous or intermittent).	
A) 2, 5, 6, 7.		
B) 1, 2, 4, 5.		
C) 2, 4, 6, 7.		
413.	Q01A	AMA
The most commo	on method of attaching a pin o	or socket to an individual wire in an MS electrical
A) crimping.		
B) soldering.		
C) crimping and	soldering.	
414.	Q01A	AMA
The pin section o	of an AN/MS connector is norr	nally installed on
A) the power sup	pply side of a circuit.	
B) the ground sid	de of a circuit.	
C) either side of	a circuit (makes no difference	).
415.	Q01A	AMA
• •	splices may be used to repair er of splices permitted betwee	r manufactured harnesses or installed wiring. The n any two connectors is
A) one.		
B) two.		
C) three.		
416.	Q01A	AMA
How should the s	splices be arranged if several	are to be located in an electrical wire bundle?
	ong the length of the bundle.	
	ther to facilitate inspection.	
C) Enclosed in a	conduit.	

<ul><li>417.</li><li>An antiskid system is</li><li>A) a hydraulic system.</li><li>B) an electrohydraulic system.</li><li>C) an electrical system.</li></ul>	R01A em.	AMA
<ul><li>418.</li><li>Antiskid braking systems ar</li><li>A) a centrifugal switch.</li><li>B) a switch in the cockpit.</li><li>C) the rotation of the wheels</li></ul>		AMA
	akeoff. The system is activated by sor.	AMA crew that a monitored flight control
A) is parallel to the longitud	angle of attack of the aircraft.	AMA when the airstream
<ul><li>421.</li><li>In an antiskid system, whee</li><li>A) an electrical sensor.</li><li>B) a discriminator.</li><li>C) a sudden rise in brake presented.</li></ul>	·	AMA
		AMA erate just below the skid point. m is turned off or if there is a system

C) both No. 1 and	a No. 2 are true.	
423.	R01A	AMA
The purpose of a	intiskid generators is to	
A) monitor hydra	ulic pressure applied to brakes	S.
B) indicate when	a tire skid occurs.	
C) measure whe	el rotational speed and any sp	eed changes.
424.	R02A	AMA
(1) A dc selsyn s movement or pos		cal method of indicating a remote mechanical
(2) A synchro typ one point to anot		trical system used for transmitting information from
Regarding the ab	oove statements,	
A) only No. 1 is to	rue.	
B) only No. 2 is to	rue.	
C) both No. 1 and	d No. 2 are true.	
425.	R02A	AMA
(Refer to Airfram	e figure 20.) What will illumina	te the amber indicator light?
A) Closing the no	sewheel gear full retract switch	:h.
B) Retarding one	throttle and closing the left w	heel gear locked down switch.
C) Closing the no	ose, left and right wheel gear f	ull retract switches.
426.	R02A	AMA
Where is the land	ding gear safety switch usually	located?
A) On the main g	ear shock strut.	
B) On the landing	g gear drag brace.	
C) On the pilot's	control pedestal.	
427.	R02A	AMA
Which repair wou	uld require a landing gear retra	action test?
A) Landing gear	safety switch.	
B) Red warning I	ight bulb.	
C) Gear downloc	k microswitch.	
428.	R02A	AMA
What landing gea	ar warning device(s) is/are inc	orporated on retractable landing gear aircraft?

<ul><li>A) A visual indicat</li></ul>	tor showing gear position.	
B) A light which co	omes on when the gear is fully o	down and locked.
C) A horn or other	r aural device and a red warning	g light.
400	DOOA	0 B 4 0
429. Tha matain an an	R02A	AMA
	tosyn remote indicating system	uses
A) an electromagr		
3) a permanent m	ragnet. etromagnet nor a permanent ma	anet
	aromagnet nor a permanent ma	griet.
430.	R02A	AMA
The rotor in a mag	gnesyn remote indicating syster	n uses
A) a permanent m	nagnet.	
3) an electromagr	net.	
C) an electromagi	net and a permanent magnet.	
431.	R02A	AMA
	e used primarily as limit switches	
A) limit generator	•	
	al units automatically.	
•	narging of a battery.	
432.	S01A	AMA
•	s in deicer boots alternately infla	
•	·	listurbance of the airflow to a minimum.
•	ion of deicer boot tubes does no	
C) Alternate inflat	ion of deicer boot tubes relieves	the load on the air pump.
433.	S01A	AMA
Why should a che	emical rain repellant not be used	on a dry windshield?
A) It will etch the (	glass.	
B) It will restrict vi	sibility.	
C) It will cause gla	ass crazing.	
434.	S01A	AMA
What is used as a	temperature sensing element i	n an electrically heated windshield?
A) Thermocouple.		
B) Thermistor.		

C) Thermometer.		
435.	S01A	AMA
What is the princip	ole of a windshield pneumation	rain removal system?
A) An air blast spr from clinging to the	•	venly over the windshield that prevents raindrops
B) An air blast for	ms a barrier that prevents rai	ndrops from striking the windshield surface.
•	nin removal system is simply matic system pressure.	a mechanical windshield wiper system that is
436.	S01A	AMA
What controls the	inflation sequence in a pneu	matic deicer boot system?
A) Shuttle valve.		
B) Vacuum pump		
C) Distributor valv	re.	
437.	S01A	AMA
What is one checl	c for proper operation of a pit	ot/static tube heater after replacement?
A) Ammeter readi	ng.	
B) Voltmeter read	ing.	
C) Continuity chec	ck of system.	
438.	S01A	AMA
	wing regulates the vacuum of cing system is off?	f the air pump to hold the deicing boots deflated when
<ul><li>A) Distributor valv</li></ul>	e.	
B) Pressure regul	ator.	
C) Suction relief v	alve.	
439.	S01A	AMA
	. •	ing by heating the leading edges of the airfoils and em usually operated during flight?
A) Continuously w	hile the aircraft is in flight.	
B) In symmetric c	ycles during icing conditions	to remove ice as it accumulates.
C) Whenever icing	g conditions are first encount	ered or expected to occur.
440.	S01A	AMA
What method is u combustion heate		e temperature of an anti icing system using surface

A) Thermo cycli	ng switches.	
B) Thermostats	in the cockpit.	
C) Heater fuel s	hutoff valves.	
441.	S01A	AMA
What is the purp	oose of the oil separator in the	pneumatic deicing system?
A) To protect the	e deicer boots from oil deterior	ation.
B) To remove oi	I from air exhausted from the	deicer boots.
C) To prevent a	n accumulation of oil in the vac	cuum system.
442.	T01A	AMA
What occurs wh	en a visual smoke detector is	activated?
A) A warning be	Il within the indicator alarms a	utomatically.
B) A lamp within	the indicator illuminates auto	matically.
C) The test lamp	o illuminates and an alarm is p	rovided automatically.
443.	T01A	AMA
Smoke detection	n instruments are classified by	their method of
A) construction.		
B) maintenance		
C) detection.		
444.	T01A	AMA
A contaminated	carbon monoxide portable tes	t unit would be returned to service by
A) heating the ir	ndicating element to 300 °F to	reactivate the chemical.
B) installing a ne	ew indicating element.	
C) evacuating th	ne indicating element with CO2	<u>)                                    </u>
445.	T02A	AMA
When used in fi	re detection systems having a	single indicator light, thermal switches are wired in
A) parallel with e	each other and in series with the	ne light.
B) series with ea	ach other and the light.	
C) series with ea	ach other and parallel with the	light.
446.	T02A	AMA
In reference to a	aircraft fire extinguishing system	ms,
(1) during remov shorted.	val or installation, the terminals	s of discharge cartridges should be grounded or

• •	o see that no voltage exists a ove statements, ue. No. 2 are true.	e electrical system, the system should be checked at the terminal connections.
447.	T02A	AMA
sensitive units tha A) parallel with ea	at complete circuits at a certa ach other, and in parallel with	-
	ach other, but in series with t	_
C) series with eac	ch other, but in parallel with t	he indicator lights.
448.	T02A	AMA
•	rea temperature is 33 °F. (R IG. IG.	pressure is acceptable for a fire extinguisher when ounded to the nearest whole number.)
449.	T02A	AMA
indicated by the a A) red disk on the B) green disk on t		that the system has been intentionally discharged is
450.	T02A	AMA
	wing are fire precautions wh	ich must be observed when working on an oxygen
1. Display 'No Sm	oking' placards.	
2. Provide adequa	ate fire fighting equipment.	
<ol><li>Keep all tools a</li></ol>	and oxygen servicing equipm	ent free from oil or grease.
4. Avoid checking	aircraft radio or electrical sy	rstems.
A) 1, 3, and 4.		

B) 1, 2, and 4.

C) 1, 2, 3, and 4.

451.	T02A	AMA
Maintenance of fire detection	n systems includes the	
A) repair of damaged sensir	ng elements.	
B) removal of excessive loop	o or element material.	
C) replacement of damaged	sensing elements.	
452.	T02A	AMA
Which fire extinguishing age	ent is considered to be the least to	xic?
A) Carbon dioxide.		
B) Bromotrifluoromethane (H	Halon 1301).	
C) Bromochloromethane (Ha	alon 1011).	
453.	T02A	AMA
The types of fire extinguishing	ng agents for aircraft interior fires a	are
•	y chemical, and halogenated hydro	
	nyl bromide, and chlorobromometh	
·	de, carbon dioxide, and dry chemic	
,	,	
454.	T02A	AMA
The proper fire extinguishing	g agent to use on an aircraft brake	fire is
A) water.		
B) carbon dioxide.		
C) dry powder chemical.		
455.	T02A	AMA
	can be checked to determine its of	
A) attaching a remote press		5 ,
B) weighing the container ar		
C) a hydrostatic test.		
, , ,		